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DOCUMENT NO. D6-58402

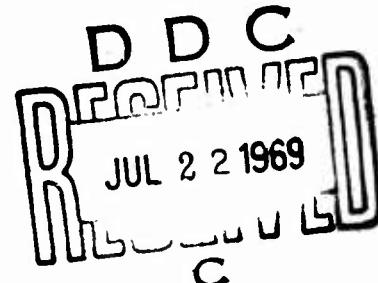
TITLE: TEMPERATURE DISTRIBUTIONS ON GREAT CIRCLE AIR ROUTES

MODEL General

ISSUE NO. 12 TO: VOC

(DATE)

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I. ABSTRACT

An accurate method of calculating the statistical monthly and annual distribution of ambient temperatures on Great Circle routes and over geographical regions is presented. An easy to operate computer program gives data in convenient graphical and tabular form for airplane system trade studies and performance calculations. () ↗

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II. SUMMARY

Accurate information concerning the statistical distribution of enroute ambient temperatures at airplane cruise altitudes is required for many trade studies and performance calculations, and is particularly important when system performance is highly temperature dependent as, for example, in airplane air-conditioning system studies. This document describes a method of calculating the statistical monthly (January, April, July and October) and annual temperature distribution on any Great Circle route for pressures of 300, 200, 150, 100 and 50 millibars, corresponding approximately to altitudes of 30,000 feet, 40,000 feet, 45,000 feet, 53,000 feet and 68,000 feet.

The route temperature distributions are generated by a computer program. Inputs to the program are the coordinates of the terminal points. Output is in several forms, namely:

1. The route temperature distribution, showing the probability of exceeding any temperature, given in graphical form with the option of tabular form as well.
2. The percentage of the total time that the temperature lies within discrete intervals of specified medium and width, given in graphical form with the option of tabular form as well.
3. The mean and standard deviation of the normal curve which best approximates the actual temperature distribution, and the error associated with the normal curve approximation.

The graphical output is illustrated in Figures 6 and 7 for the Johannesburg to London route at 30,000 feet.

Meteorological data based on records compiled over long periods for a selected global network of points furnish the basis for the program.

Sections of an existing program (Boeing Document D6-6833TN, Program No. TAPO03) are employed, as a subroutine, for the determination of mean temperatures and standard deviations at equidistant points 100-200

nautical miles apart along a Great Circle route. The results were then used in the earlier program to generate the route mean temperature and the mean value of the enroute standard deviations for many routes, as described in documents D6-15650-2 and D6-15650-4. Presentation of results in this form has the disadvantage that the averaging process obscures the effect of temperature variations along a route. Furthermore, representation of the temperature distribution by a normal curve results in a loss in accuracy. For some cases, and particularly for short East-West routes and for all routes at 40,000 feet, enroute temperature variations are small and the route-mean results will closely approximate the results developed in this document. However in many cases enroute temperature variations are significant, and misleading information about temperature extremes will be obtained if the route-mean results are employed to describe the enroute temperature distribution.

Temperature distributions for specific routes may prove unnecessary voluminous to a user who only requires information for a specific region of the earth. Therefore results are presented for four geographical regions of interest, namely:

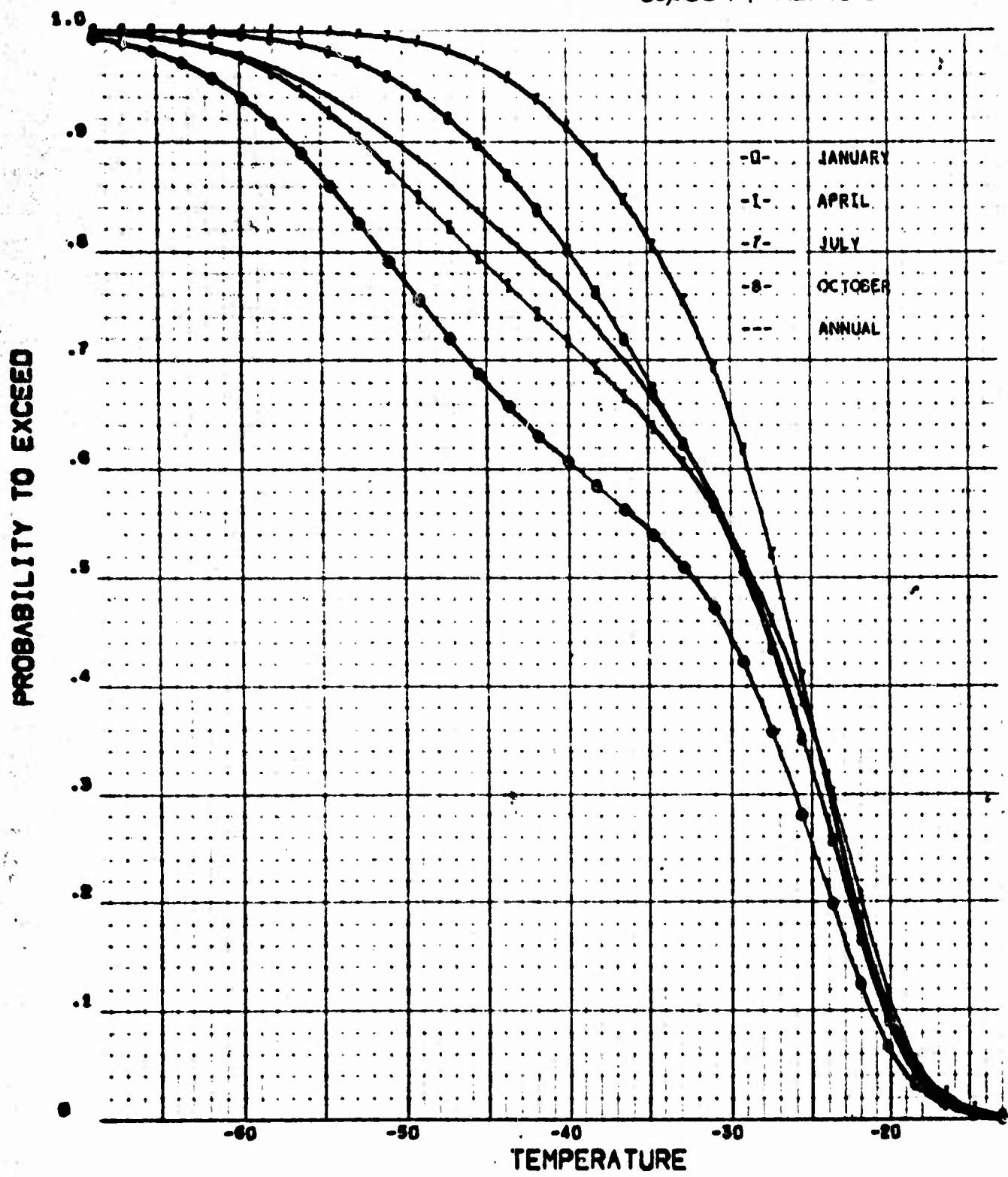
1. The continental United States.
2. The continental United States combined with the North Atlantic and Central and Southern Europe.
3. The polar region of the Northern Hemisphere.
4. The equatorial region.

Temperature distributions for these regions are compared with the U. S. Standard Atmospheres, and it is shown that the Hot and Cold Atmospheres are encountered only over limited and often remotely located areas.

The advice and cooperation of Mr. Neal M. Barr (Atmospheric Engineering Group, Aerodynamic Staff) is acknowledged.

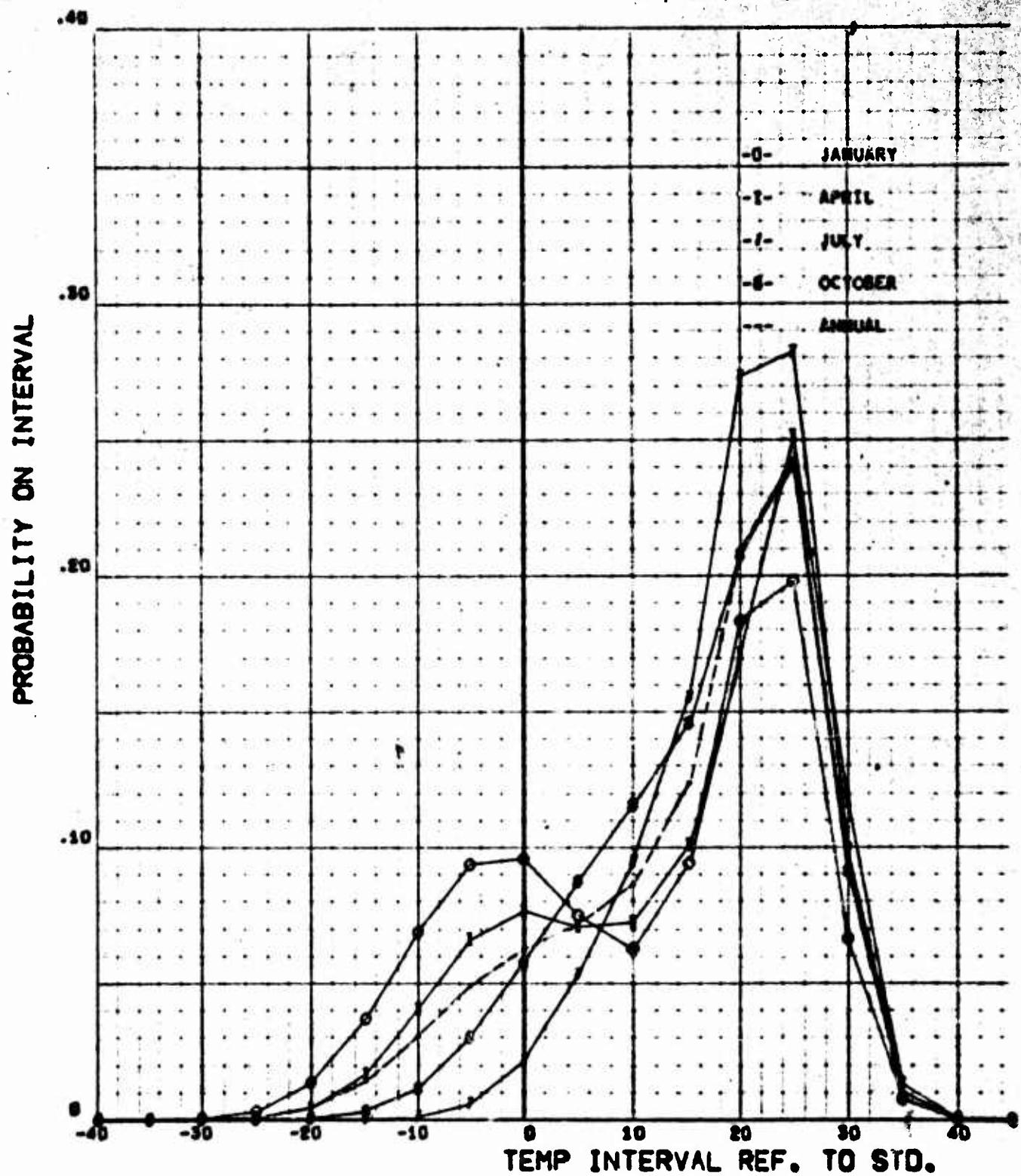


JOHANNESBURG TO LONDON
30,000 FT ALTITUDE



CASE			RE-140	DATE	SEASONAL AND ANNUAL DISTR. FOR GIVEN ALT. AND ROUTE	FIG. 6
CHECK						
APPN						D6-58402
FLYING						Page 6

JOHANNESBURG TO LONDON
30,000 FT ALTITUDE



CALC			AER SEC	DATE	PROBABILITY ON INTERVALS OF SPECIFIED WIDTH	FIG. 7 D6-58402
CHECK						
APP'D						
ALL						

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III. INTRODUCTION

The ambient air temperature influences the performance of many airplane systems. Accurate information about this temperature is, therefore, of prime importance not only for trade studies but also for performance calculations.

Many different methods of estimating representative ambient temperature distributions have been used. One approach is to consider just the defined values of Standard, Hot and Cold Atmospheres, and to assume that these occur in fixed proportions (90 percent, 10 percent, 0 percent, for example). Such an approach is clearly very arbitrary and is unlikely to yield accurate results. Some of the shortcomings of this method are avoided by using the route-mean temperature data which have been compiled for many Great Circle routes (Reference 1). Route data in Reference 1 are given in the form of the route mean temperature, the mean value of the enroute standard deviations, and the temperatures that are not exceeded 50 percent, 75 percent and 85 percent of the time. The effect of inflight temperature variations is not reflected in these data and therefore information about temperature extremes calculated from the given mean temperature and standard deviation values, may be misleading.

The method of calculation described in this document avoids the approximations introduced in employing the route-mean data of Reference 1 to determine the enroute temperature distribution of a studied route, and the error involved in approximating the actual distribution by a normal curve. Both the developed method and the route-mean method use the same meteorological data. The deviation in the information about temperature extremes between the two methods is dependent on the magnitude of the enroute temperature variation, and this is in turn a complex function of the route, the altitude and the time of the year. In general, the difference increases with length of route, is greater for North-South than East-West routes, and is comparatively high at an altitude of 53,000 feet and low at 40,000 feet.

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IV. CALCULATION OF TEMPERATURE DISTRIBUTION ON A GREAT CIRCLE ROUTE

A. METEOROLOGICAL DATA

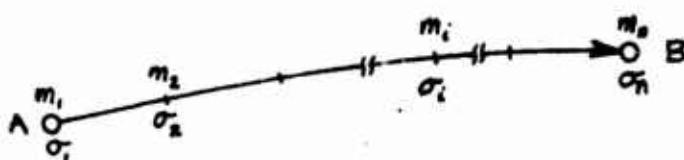
Meteorological data, based on records compiled over long periods, have been obtained for a network of 1117 points covering the surface of the earth. The points are located at every latitude which is an exact multiple of five degrees. Within 60° latitude of the equator, the points are located at every longitude which is an exact multiple of ten degrees; on those latitudes which are farther than 60° from the equator the points are located at every longitude which is an exact multiple of twenty degrees. Each pole is represented by one point. The mean and standard deviation of a normal distribution fitted to the actual temperatures over a period of a month are recorded for each point. Data are available for January, April, July and October, these months being assumed to be representative of the seasons; and for pressures of 300, 200, 150, 100 and 50 millibar corresponding approximately to altitudes of 30,000 feet, 40,000 feet, 45,000 feet, 53,000 feet and 68,000 feet. Thus a total of 44,680 data values are available.

B. GREAT CIRCLE ROUTE CALCULATION

If the geographical coordinates of the terminals of a route are specified, the Great Circle route may be determined by standard methods (Reference 2 for example). The coordinates of most major airports can be found in Reference 1.

C. CALCULATION OF ROUTE TEMPERATURE DISTRIBUTION

If a route is divided into a number of equidistant points, the mean temperature (m_i), and standard deviation (σ_i) at each point (for a given month and altitude) may be obtained by interpolation from adjacent data points (Reference 2). Then if the temperature at any



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point (i) is distributed normally, the probability that the temperature T at this point is less than or equal to any specified temperature ϑ is, by definition:

$$P(T \leq \vartheta) = F_N(\vartheta)_i = \frac{1}{\sqrt{2\pi}\sigma_i} \int_{-\infty}^{\vartheta} e^{-\left(\frac{T-m_i}{2\sigma_i}\right)^2} dT \quad (1)$$

The probability that the temperature ϑ is exceeded is given by

$$P(T > \vartheta) = 1 - F_N(\vartheta)_i \quad (2)$$

If the distance between points is chosen so that the temperature and standard deviation can be assumed constant during each flight increment, then the probability for the whole route of any specified temperature ϑ being exceeded is given by

$$P_R(T > \vartheta) = 1 - F_R(\vartheta) = \frac{\sum_{i=1}^n (1 - F_N(\vartheta)_i)}{n} \quad (3)$$

An example of this method, using hand calculations to determine the cumulative probabilities, is shown in figures 1 and 2. In order to limit the number of calculations, the example uses larger temperature steps and fewer significant figures. The difference between the temperature distribution at selected points and the route temperature distribution is illustrated in figure 2.

The probability of the temperature falling within a specified interval is given by:

$$P(\vartheta < T \leq \vartheta + \Delta\vartheta) = F_R(\vartheta + \Delta\vartheta) - F_R(\vartheta) \quad (4)$$

Assuming that the months of January, April, July and October are representative of the four seasons, the annual probabilities may be obtained. Thus:

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$$P_{RA}(T > \vartheta) = 1 - F_{RA}(\vartheta) = \frac{\sum_{\vartheta=1}^{\infty} (1 - F_R(\vartheta))}{4} \quad (5)$$

$$P_{RA}(\vartheta < T \leq \vartheta + \Delta \vartheta) = F_{RA}(\vartheta + \Delta \vartheta) - F_{RA}(\vartheta) \quad (6)$$

D. FITTING OF A NORMAL CURVE

For most routes the temperature distribution closely approximates a normal curve. Since normal curves may be defined by specifying just the mean temperature and standard deviation, it is worthwhile determining the normal curve most closely approximating the actual distribution.

The mean temperature of the normal curve is calculated from the actual temperature which is exceeded 50% of the time and from temperatures having the defined probabilities of $\pm \sigma/2$ and $\pm \sigma$, thus averaging out discrepancies in any single value and giving:

$$m_F = \frac{1}{5} (\vartheta_{-\sigma} + \vartheta_{-\sigma/2} + \vartheta_{50} + \vartheta_{\sigma/2} + \vartheta_{\sigma}) \quad (7)$$

The standard deviation is found by a similar method, averaging values at $\pm \sigma/2$, $\pm \sigma$, ..., $\pm 3\sigma$, and giving:

$$\sigma_F = \frac{1}{12} \left[\sum_{2\sigma=-6}^{2\sigma=4} ((\vartheta_{\gamma} - m_F)/\gamma) + \sum_{2\sigma=1}^{2\sigma=6} ((\vartheta_{\gamma} - m_F)/\gamma) \right] \quad (8)$$

The mean and maximum deviations in $(1 - F_R(\vartheta))$ between the actual distribution and the fitted curve are calculated in order to indicate the accuracy of the curve-fitting.

E. CALCULATION OF A PROBABLE REGIONAL TEMPERATURE DISTRIBUTION

The available meteorological data can also be utilized to generate a probable temperature distribution that would apply to a geographical region of the earth rather than individual routes. The less voluminous results applicable to an airplane typically operating over the specified region are obtained at the expense of averaging

the temperature profiles over the region.

A method of calculation similar to Paragraph IV, C is used to generate the regional temperature distribution based on the meteorological data points located within the specified region. The spheroid shape of the earth is accounted for by giving area-balanced weights to individual points. Results are given in Appendix A.

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V. DESCRIPTION OF PROGRAM

A complete listing of the computer program for calculating enroute temperature distributions is given in Appendix B. It can be seen from the listing that the program is divided into a main program, five subroutines and a function subprogram. The main program handles input and output routines, and directs control to the subroutines and the function subprogram. The meteorological data is stored on a tape-unit (called TAPEx for cross-reference) assigned to the author. The subprograms have the following functions:

1. Subroutine TEMP.

Sections of an existing program (ref. 2) are used for generating mean temperatures and standard deviations at equidistant points 100-200 nautical miles apart along the specified route.

2. Subroutine DISTR.

Determines the statistical temperature distribution for the whole route from the mean and standard deviation data for the individual points.

3. Subroutine FITNC.

Determines the mean and standard deviation of the normal curve most closely approximating the calculated distribution curve.

4. Subroutine PINT.

Determines the probability of the temperature being within intervals of given median and width. The system of median temperatures are centered around the Standard Atmosphere temperature.

5. Subroutine WBIN

Prepared calculated data for input to the CDC-6600 system tape-writing subroutine WRTETP (ref. 4).

6. Function subprogram CNORML

This subprogram calculates the cumulative probability for a normal distribution, given the mean and standard deviation and the region for which the probability is required (ref. document PS-497, Program Library Department).

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Input for the program is described in Appendix C, and sample data for the route Johannesburg to London are included for illustration. It should be noted that exactly antipodal points must not be specified as terminals of a route since such points do not give a uniquely defined Great Circle route. Furthermore, meteorological data are not available on the 50 millibar (68,000' altitude) level south of latitude 55 S. A diagnostic is printed if the Great Circle route enters this region.

In order to reduce core storage requirements the **DIMENSION** statements have been written so that not more than two altitudes can be handled in any one run, and routes are restricted to a maximum length of 11,000 nautical miles.

Output data are in the form of graphs and tables and are self-explanatory. The plots are not labeled by route, but may be identified by the fact that they are generated in the same order that the routes are inputted. Tabular and graphical output corresponding to the input of Appendix C are shown in figures 5 through 7.

Appendix D shows a complete deck assembly and lists the required control cards. The Fortran nomenclature is shown in Appendix E. The core storage requirement is 110000₈ and the central processor time is approximately equal to

$$CPT = 21.5 + 3.8 \times NALT \times NROUTE \text{ seconds}$$

VI. EXAMPLE

A. ROUTE TEMPERATURE DISTRIBUTION

Use of the program may be illustrated by considering a specific route, and Johannesburg to London at 30,000' has been selected. As this is a long route (4896 N.M.) and in a South-North direction, considerable enroute temperature variations can be expected. This is confirmed by reference to figure 3, which shows isotherms at 30,000' in January. Figure 4 shows that the standard deviations varies noticeably too.

The mean temperature and standard deviation for 26 equispaced points along the route are shown in figure 5, which also shows temperatures having a 50, 75 and 85% probability of not being exceeded. These values could also be obtained from the graphical plot of figure 6, and can be compared with values obtained using the route-mean method of reference 1. For the month of January, the comparison shows:

<u>Probability of Not Exceeding</u>	<u>Enroute Temperature Method</u>	<u>Route-Mean Method</u>
50%	-32.3F	-38 C \approx -36F
75%	-24.9F	-37 C \approx -35F
85%	-22.6F	-36 C \approx -33F

Figure 5 shows the mean and standard deviation of the normal distribution curve most closely resembling the actual temperature distribution, and also the error associated with the normal curve approximation. It can be seen that in this case, the normal curve is not a particularly good approximation. A comparison for the month of January with the values listed in reference 1 gives:

	<u>Enroute Temperature Method</u>	<u>Route-Mean Method</u>
Mean Temperature	-36.1F	-38C \approx -36F
Standard Deviation	-13.4F	2C \approx 4F

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It can be seen that the route-mean method gives misleading information about the extremes of temperature that will be encountered on this route and this method, originally not intended to describe the enroute temperature distribution but frequently used in this capacity, should not be used where considerable enroute temperature variations can be expected.

It is interesting to apply the different methods of determining route temperatures to the calculation of the operating penalty of a system. Consider, as an example, the ram drag of the model 747 air conditioning system at .85 MN cruise. The ram drag is plotted in Figure 8 as a function of ambient temperature, with temperature histograms based on the statistical and route-mean methods superposed. The average route ram drag may be compared, giving:

<u>Method</u>	<u>Average Drag lb/Airplane</u>
Enroute temperature	799
Route-Mean	396
Standard Day 90% }	
Hot Day 10% }	415

It can be seen that in this case the route-mean and Standard/Hot day methods seriously underestimate the average drag, and hence the average airplane penalty.

B. REGIONAL TEMPERATURE DISTRIBUTION

Appendix A introduces the use of the available meteorological data to generate a probable temperature distribution applicable to a specified region of the earth, thereby eliminating the high degree of idealization inherent to the use of standardized atmospheres. To illustrate the use of the generated data, an example involving the identical airplane-system as in the previous example will be presented. Two of the selected geographical regions in Appendix A (Continental United States and Equatorial region) at the 30,000 foot altitude will be studied.

From the annual temperature distribution curves for the two regions at the given altitude (Figures 14 and 17) one can easily obtain the

probabilities on 5 degrees Fahrenheit intervals for calculation of the average ram drag, as was done for an individual route in the previous section. Figure 10 gives the probability histograms for the two regions superposed on the ram drag curve. Comparison of the average ram drag gives:

	<u>Average drag lb/airplane</u>
Standard Day 90% }	415
Hot Day 10% }	
Probable temp. distribution over the Continental United States	447
Probable temp. distribution over the Equatorial region	1260

It is seen that for the given example, the adopted combinations of the standardized atmospheres yields an acceptable value of the average drag if applied over the Continental United States. If another airplane system, having a different temperature dependence, was being studied, this might not have been the case. Taken over the equatorial region it is found that 90% Standard Day, 10% Hot Day seriously under-estimates the average drag, in spite of the fact that Hot Day (Hot Day = Std. Day + 40F at 30,000 ft.) is found to be exceeded < 1% of the year when focusing the entire equatorial region of the earth. Likewise it can be seen that temperatures as low as Standard Day are extremely unlikely to be encountered over this area, making an assumption of 90% of the time at Standard Day unrealistic. Finally if the interest is founded on localized areas where the concept of Hot Day applies (areas recording temperatures exceeding Hot Day 10% of the time), one will find that such an area at 30,000 feet exists over South East Asia. Calculations of the average drag based on the probable temperature distribution over this area will result in a value exceeding the one obtained for the Equatorial region, showing that if 10% Hot Day is a realistic figure, one can not designate Standard Day condition to the remaining portion of the time.

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VII. CONCLUSIONS AND RECOMMENDATIONS

The method for determining enroute temperatures developed in this document generates results which are limited in accuracy only by the accuracy of the meteorological data on which they are based. The computer program is simple to use and presents results in convenient tabular and graphical forms. Therefore, it is recommended that the statistical method should always be used for generating enroute temperature distributions. For some routes, and particularly for short East-West routes, the route-mean method (Reference 1) gives acceptable approximations of the enroute temperature distribution.

Use of the regional temperature data to designate a probable temperature distribution eliminates the high degree of idealization inherent to the use of standardized atmospheres, thereby greatly improving the accuracy and reliability of employed data. It is recommended that the regional temperature data be used instead of standardized atmospheres in determining typical operating temperatures for airplanes and airplane systems.

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VIII. FIGURES

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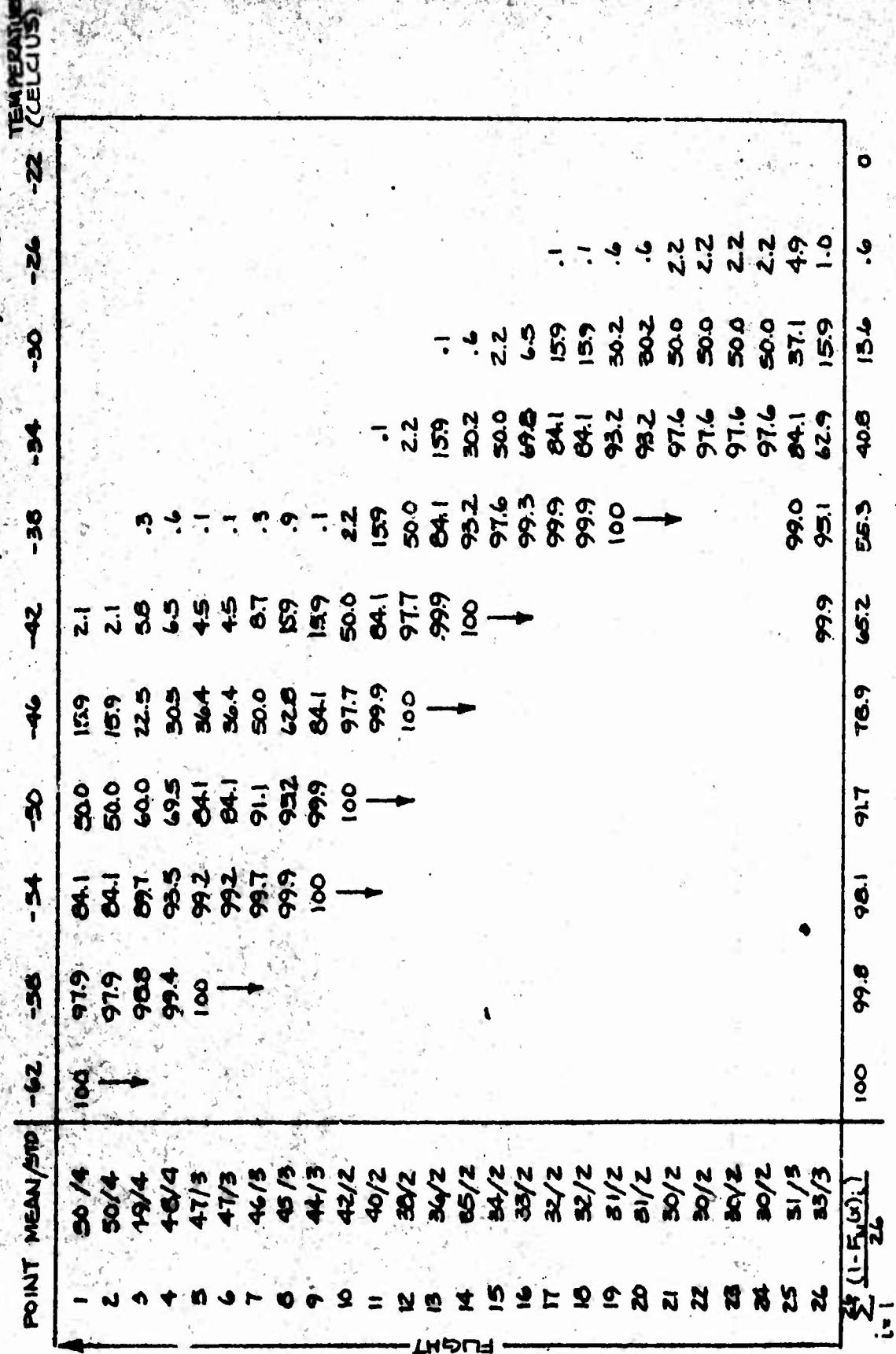
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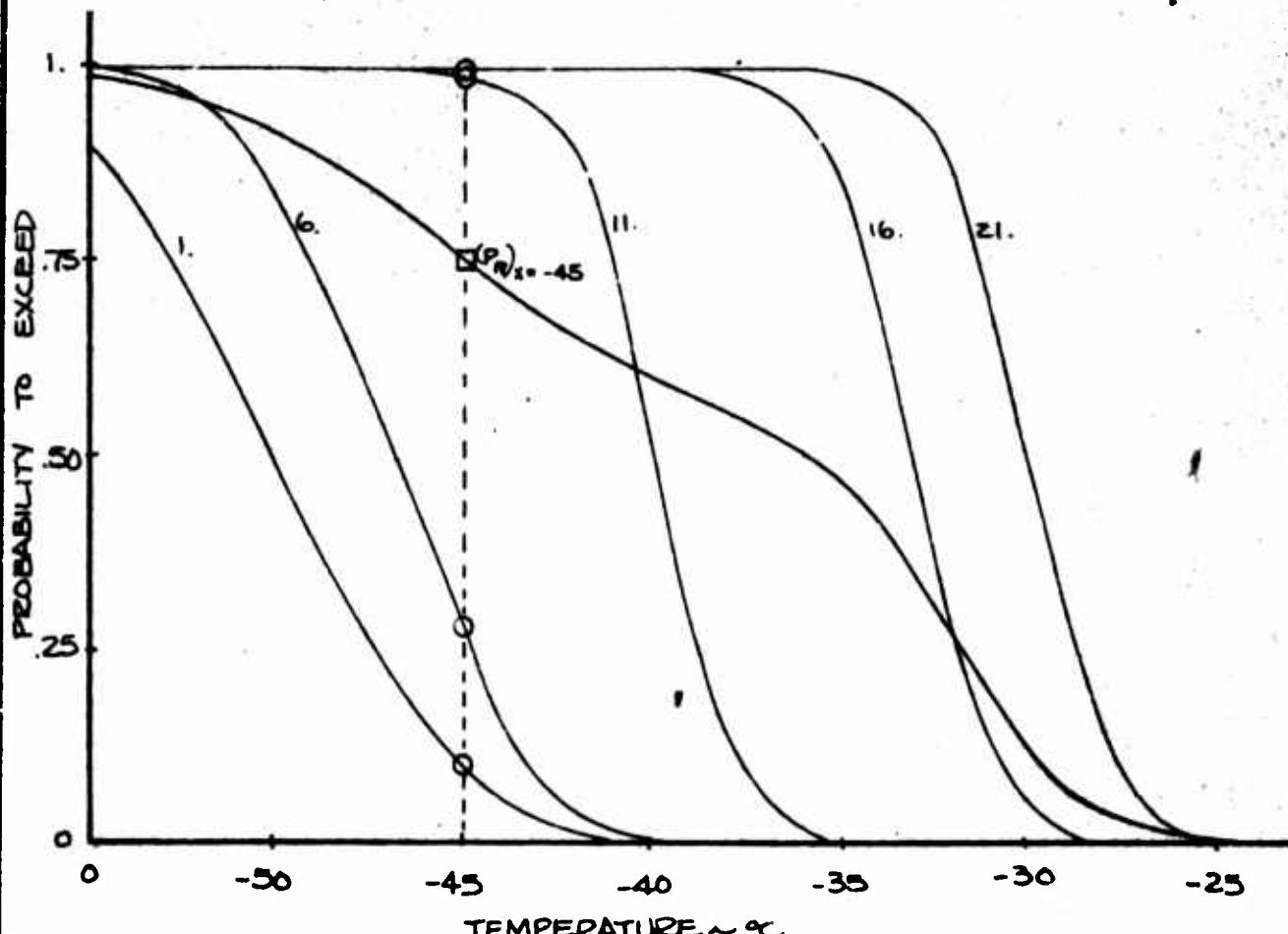
PROBABILITY TO EXCEED

JOHANNESBURG TO LONDON
30,000 FT ALTITUDE, JUNE 1966 N.M.

CALC			REVISED	DATE	HAND CALCULATION OF ROUTE TEMPERATURE DISTRIBUTION	FIG. 1
CHECK						DD-58402
APPD						PAGE 20
APPD					THE BOEING COMPANY RENTON, WASHINGTON	
Dan	Dub					6-1000

JOHANNESBURG TO LONDON
30,000 FT, JANUARY
CONT.

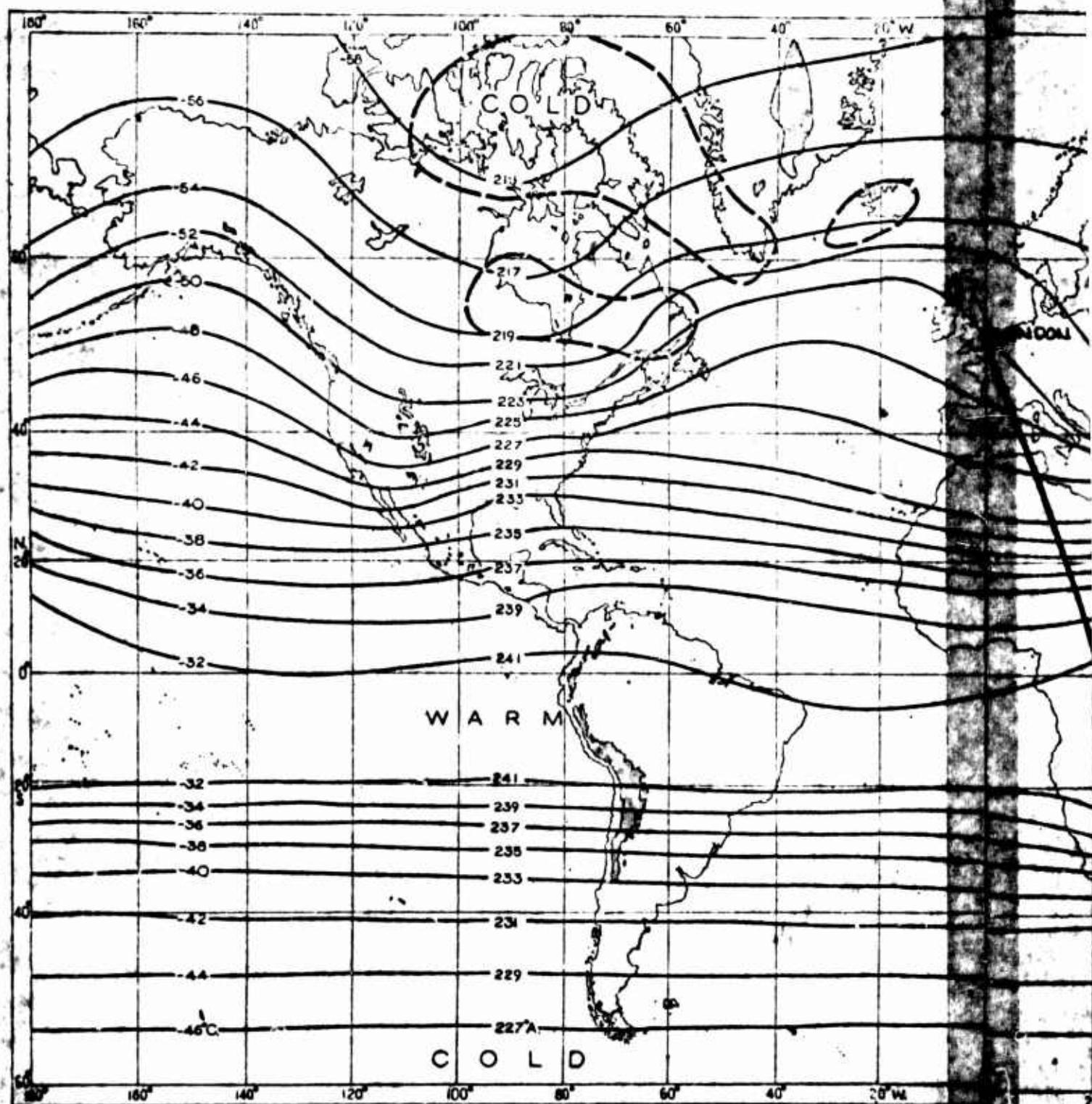
THE NORMAL DISTRIBUTION CURVE FOR A NUMBER OF SELECTED POINTS IS SHOWN BELOW. THE CURVE (NON-NORMAL) BASED ON THE CALCULATED DATA IN FIGURE GIVES THE DISTRIBUTION FOR THE ENTIRE ROUTE.



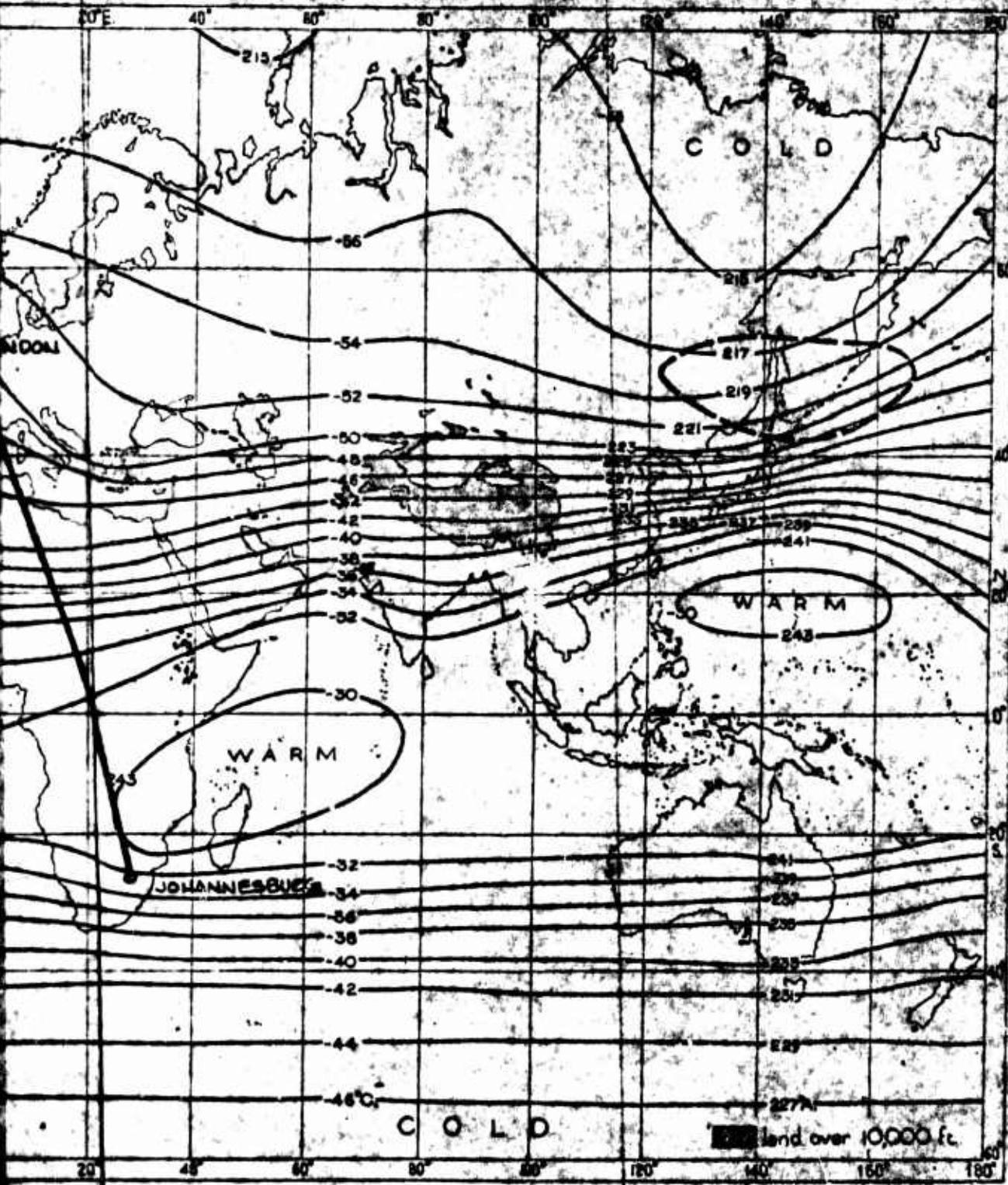
EXAMPLE: $(P_R)_{x=-45} = \frac{\sum (P_i)_{x=-45}}{26} = 75.4\%$

CALC			REVISED	DATE	HAND CALCULATED ROUTE TEMPERATURE DISTRIBUTION CURVES	FIG. 2
CHECK						D6-58402
APPD						21
APPD						
DR	Aero					

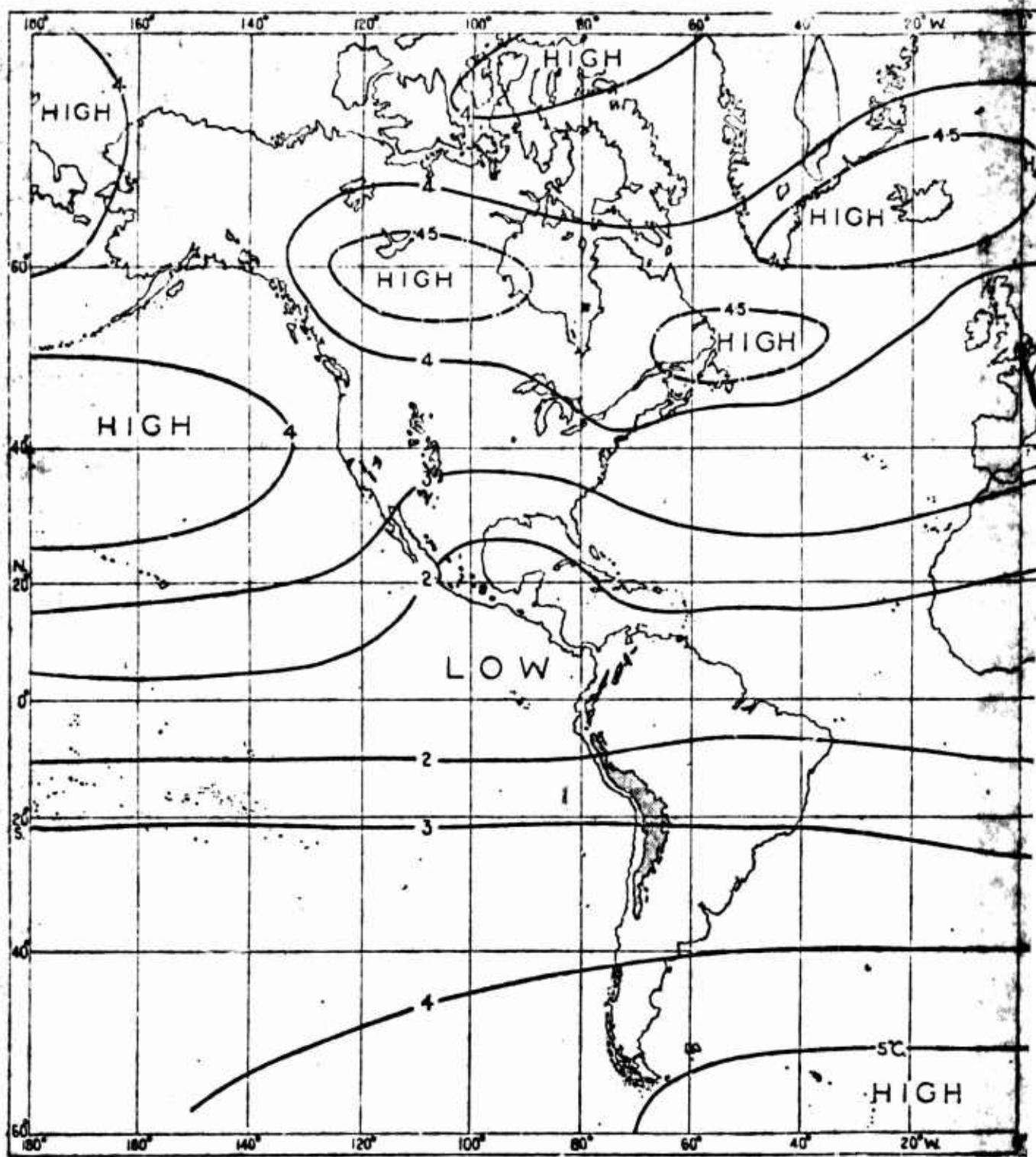
THE BOEING COMPANY
RENTON, WASHINGTON



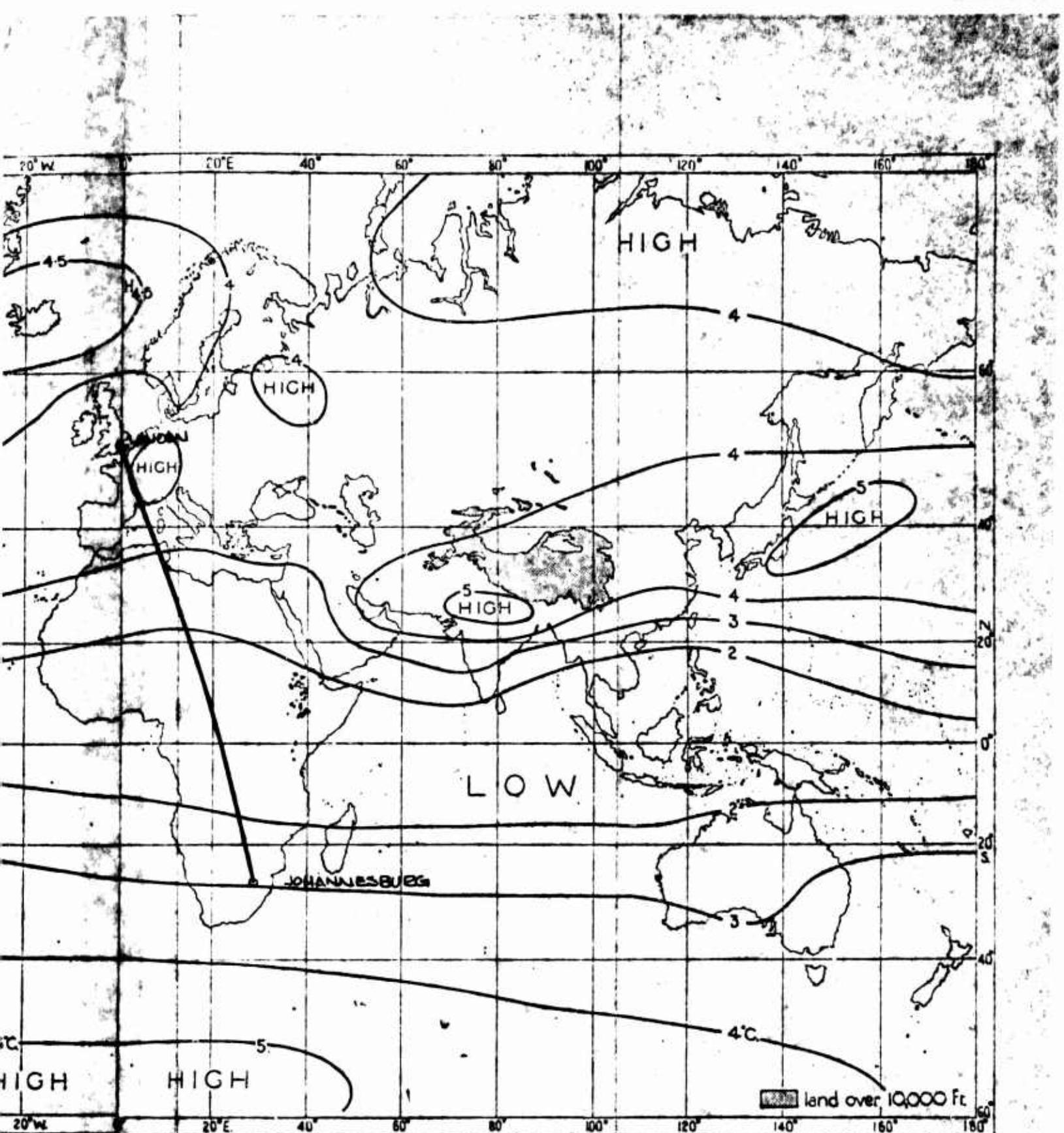
AVERAGE TEMPERATURE AT 300 MB. IN JANUARY
UPPER AIR TEMPERATURE OVER THE WORLD
N. GOLDIE, J.G. MOORE AND E.E. AUSTIN



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Fig. 3



STANDARD DEVIATION OF TEMPERATURE AT 300 MB. IN JANUARY
UPPER AIR TEMPERATURE OVER THE WORLD
N. GOLDE, J.G. MOORE AND E.E. AUSTIN



N JANUARY

B

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FIG. 4

CASE NO. 1 OUT OF NUMBER 1

JOHANNESBURG TO LONDON 900A N.M.

START POINT	LATITUDE	•	-37°40.88'
	LONGITUDE	•	-78°11.13'
END POINT	LATITUDE	•	51°24.84'
	LONGITUDE	•	-71°11.07'

ALTITUDE 3000FT

MEAN/ST.DEV. FOR POINTS ALONG THE ROUTE	JANUARY	MARCH	JULY	OCTOBER
DIST M.M.				
0	-32.7/2.4	-31.9/3.1	-37.8/3.5	-36.6/3.2
196	-31.3/2.6	-30.5/2.9	-35.7/3.1	-35.0/3.0
392	-30.2/2.3	-32.9/2.7	-33.9/2.8	-32.6/2.6
588	-30.2/2.1	-31.1/2.5	-33.6/2.6	-32.1/2.4
783	-30.2/2.0	-31.3/2.4	-33.7/2.3	-31.6/2.2
974	-30.6/1.9	-30.7/2.2	-33.8/2.2	-31.0/2.1
1175	-30.7/1.7	-30.8/2.0	-32.6/2.0	-30.6/2.0
1371	-31.0/1.6	-29.7/1.6	-32.3/1.9	-29.6/1.6
1567	-31.5/1.6	-29.9/1.7	-32.1/1.8	-29.6/1.6
1763	-37.2/1.6	-29.6/1.7	-31.7/1.8	-29.9/1.7
1959	-37.9/1.6	-29.5/1.6	-31.6/1.7	-30.6/1.7
2156	-33.7/1.6	-29.6/1.6	-31.1/1.6	-31.0/1.7
2350	-34.7/1.6	-30.3/1.6	-31.7/1.6	-31.6/1.6
2546	-34.6/1.7	-31.2/1.7	-30.3/1.9	-32.6/1.8
2742	-37.7/1.6	-32.6/1.7	-29.9/2.0	-32.6/1.8
2938	-39.6/2.0	-33.8/1.9	-29.4/2.1	-33.6/1.9
3133	-42.1/2.1	-35.6/2.1	-29.5/2.3	-35.6/2.1
3329	-43.0/2.0	-37.6/2.3	-30.3/2.5	-36.6/2.1
3525	-44.9/2.7	-40.2/2.5	-31.8/2.8	-36.6/2.3
3721	-45.8/2.4	-42.6/2.9	-33.4/3.0	-39.6/2.8
3917	-46.7/2.2	-44.4/3.3	-35.5/3.1	-40.9/3.3
4113	-47.6/3.4	-45.6/3.6	-36.6/3.2	-41.1/3.9
4309	-44.3/3.0	-44.1/3.0	-37.7/3.4	-41.0/3.1
4505	-46.1/4.0	-46.1/4.2	-34.8/3.5	-42.6/4.2
4700	-49.5/3.9	-47.1/3.9	-34.8/3.6	-43.3/4.0
4896	-49.9/3.9	-47.3/3.7	-40.7/3.7	-44.0/4.0

INFORMATION UNIT IN FOLLOWING TABLES IS **ONE PARENTHETIC ONE**

TEMP. THAT CORRESPONDS TO GIVEN PHUM. & O T IN HF RECEIVED	ANNUAL				
PHUM. VALUE	JANUARY	APRIL	JULY	OCTOBER	
.50	-32.3	-24.7	-27.1	-24.0	-24.0
.75	-26.9	-22.9	-23.5	-23.7	-23.6
.95	-22.6	-20.4	-21.1	-21.0	-21.5

SEE A FANTASTIC CHANCE TO WIN MONEY DURING

	JANUARY	APRIL	JULY	OCTOBER	ANNUAL
AVERAGE TEMPS.	-10.11	-32.00	-21.45	-10.61	-31.40
ST. DEG. &	13.33	11.70	9.30	9.61	10.06

NEARNESS OF FITTED CURVE TO MEASURED DISTRIBUTION
DEVIATION IN PROBABILITY VALUES FOR CALC. TEMPERATURES

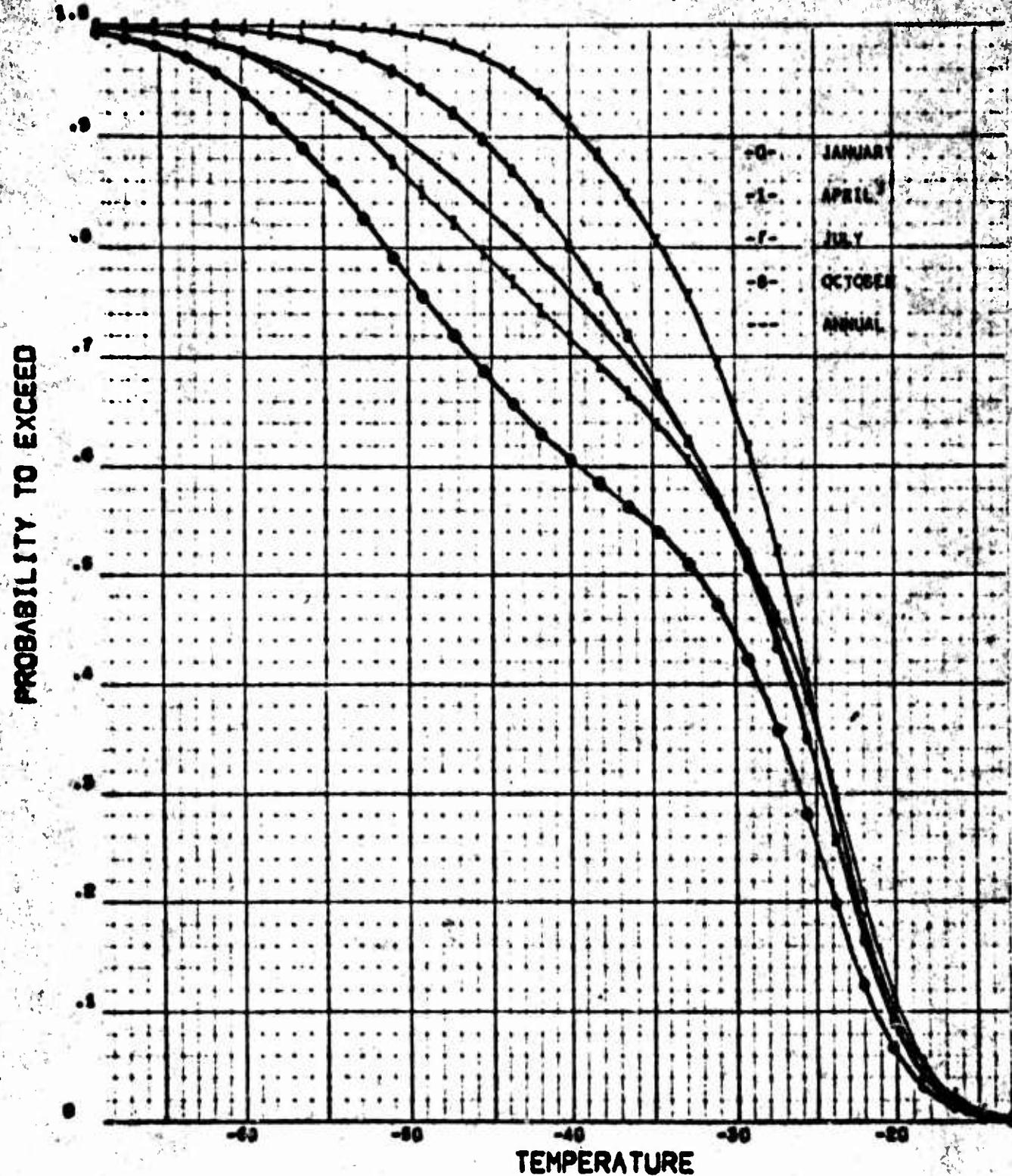
MEAN, MEDIAN, DEVIATION, MAXIMUM, MINIMUM

Figure 5

D6-58402

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JOHANNESBURG TO LONDON
30,000 FT ALTITUDE

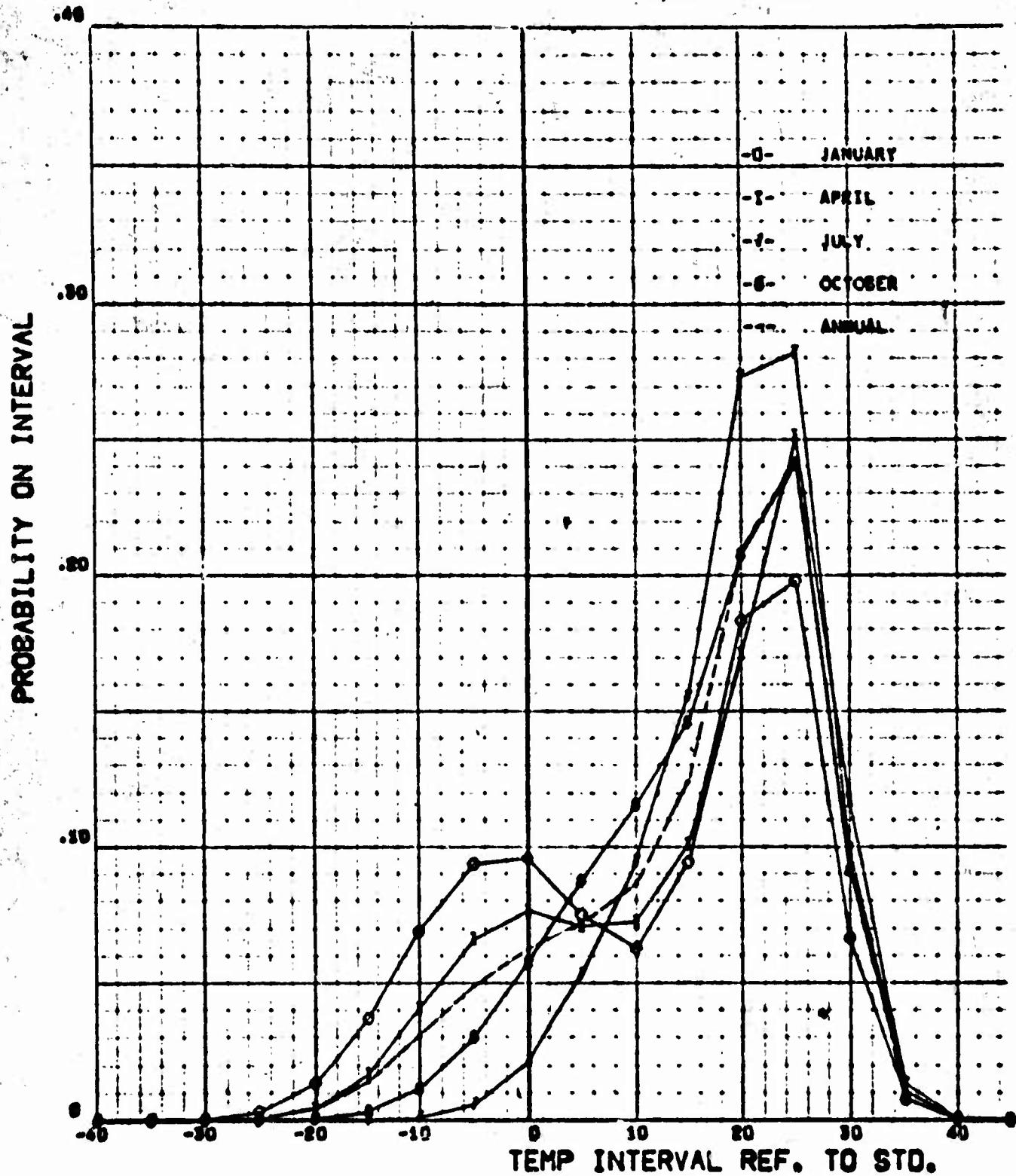


CALC			BL. 102	24.1	SEASONAL AND ANNUAL DISTR. FOR GIVEN ALT. AND ROUTE	FIG. 6
SOURCE						D6-58402
APR.						
OCT.						

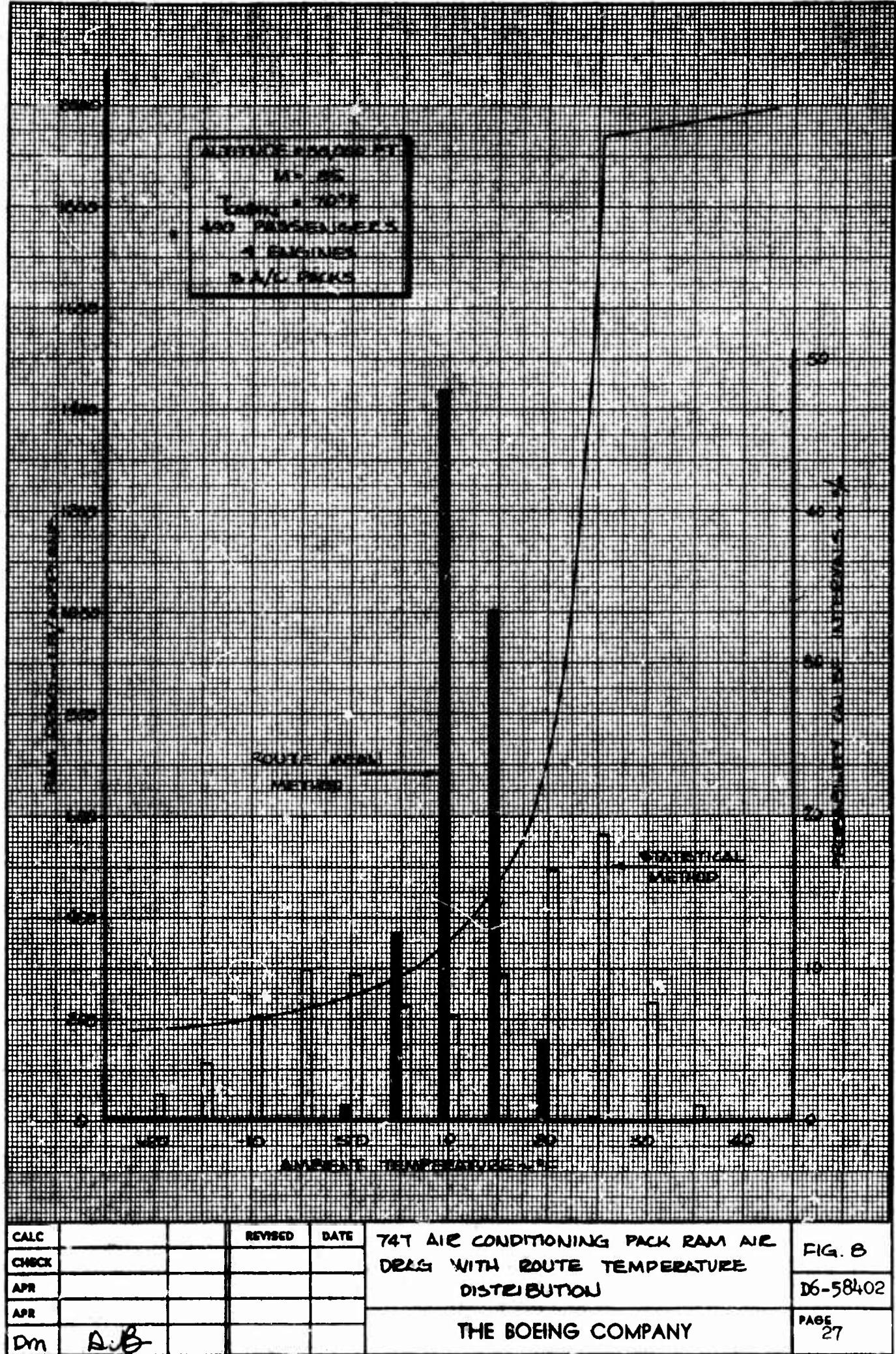
THE BOEING COMPANY

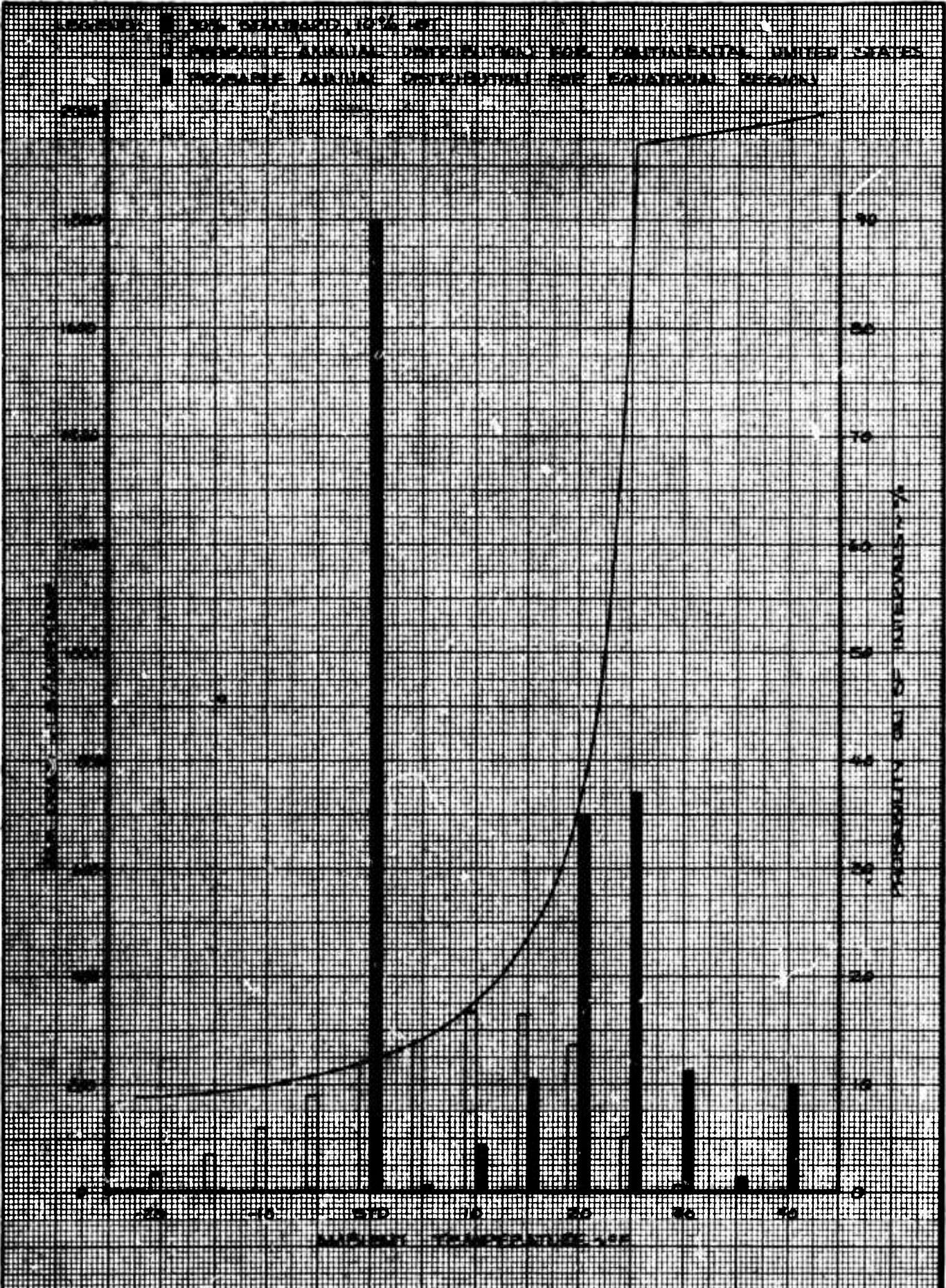
25

JOHANNESBURG TO LONDON
30,000 FT ALTITUDE



SAC		REF ID: 5411	PROBABILITY ON INTERVALS OF SPECIFIED WIDTH	FIG. 7
SOURCE				D6-58402
APPD				
AFPD			BOEING	26





CALC			REVISED	DATE
CHECK				
APR				
APR				
Dra	<i>Dub</i>			

747 AIR CONDITIONING PACK RAM AIR
DRAG WITH REGIONAL TEMPERATURE
DISTRIBUTION

THE BOEING COMPANY

FIG. 9

DB-58402

PAGE
28

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2. Document D6-6833TN, Enroute Temperatures, L. A. Rasmussen, Dec. 1963.
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4. Document D6-19175TN, SC-4020 Plot Package, David D. Renhed, Dec. 1965.
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6. MIL STD -210A. Climatic Extremes for Military Equipment.
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AD 1546 D

REV SYM

BOEING

NO. D6-58402

PAGE 29



6-7000

APPENDIX A
USE OF THE AVAILABLE METEOROLOGICAL DATA
TO GENERATE REGIONAL TEMPERATURE DISTRIBUTIONS

The concepts of Standard, Hot and Cold Atmosphere (References 5 through 7) are commonly combined to designate a probable temperature distribution for trade studies or performance calculations on the typical operation condition of an airplane system being studied. Any such combination of the standardized atmospheres (i.e., 90 percent of the time at Standard Day and remaining 10 percent at Hot Day) is likely to involve a substantial degree of idealization of any probable temperature distribution and the risk for misinterpretations becomes apparent. If Standard Day condition is encountered during the major part of the time, as in above example, the probability of experiencing Hot Day is extremely remote. On the other hand, if the interest is focused on the localized areas where Hot Day temperature actually is being exceeded the specified percentage of the time, it is improbable that temperatures as low as Standard Day will be encountered.

Using the available meteorological data, it is possible to generate a more realistic temperature distribution curve applicable to a specified region of the earth, thereby eliminating the uncertainties inherent in the use of the standardized atmospheres. The developed data becomes applicable to an airplane in traffic of about equal density over the specified region. Data for four regions of common interest has been generated. These are:

1. Continental United States with adjacent border areas (52.5°N to 22.5°N latitude and 65°W to 125°W longitude).
2. Area 1 combined with North Atlantic, Central and Southern Europe (Area 1 plus 62.5°N to 37.5°N latitude and 25°E to 65°W longitude).
3. Polar region of Northern Hemisphere (90°N to 62.5°N latitude).
4. Equatorial region (22.5°N to 22.5°S latitude).

Data pertinent to any other region of the earth of interest can, upon request, be generated.

AO 1948

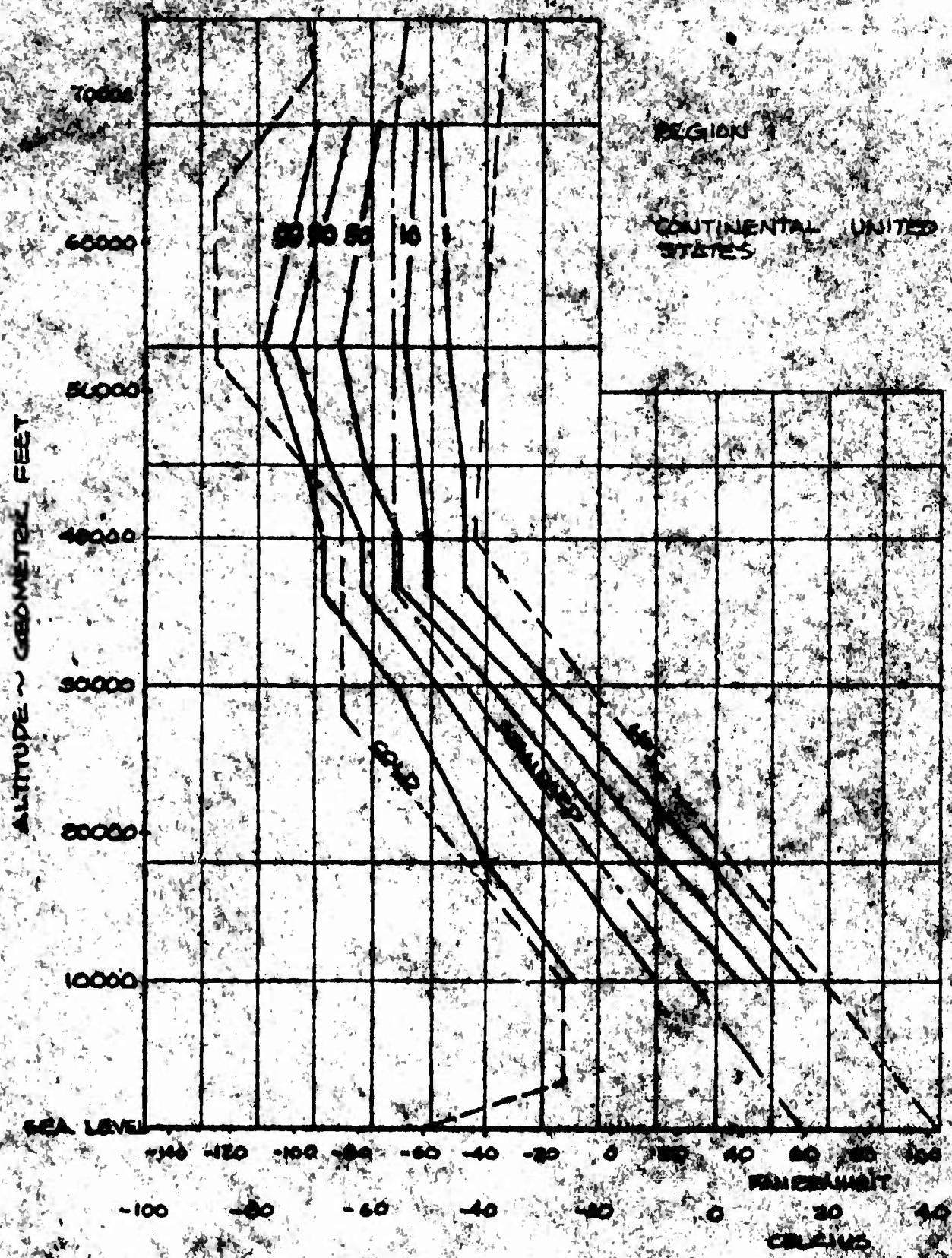
REV SYM

BOEING NO. D6-58402
PAGE 30



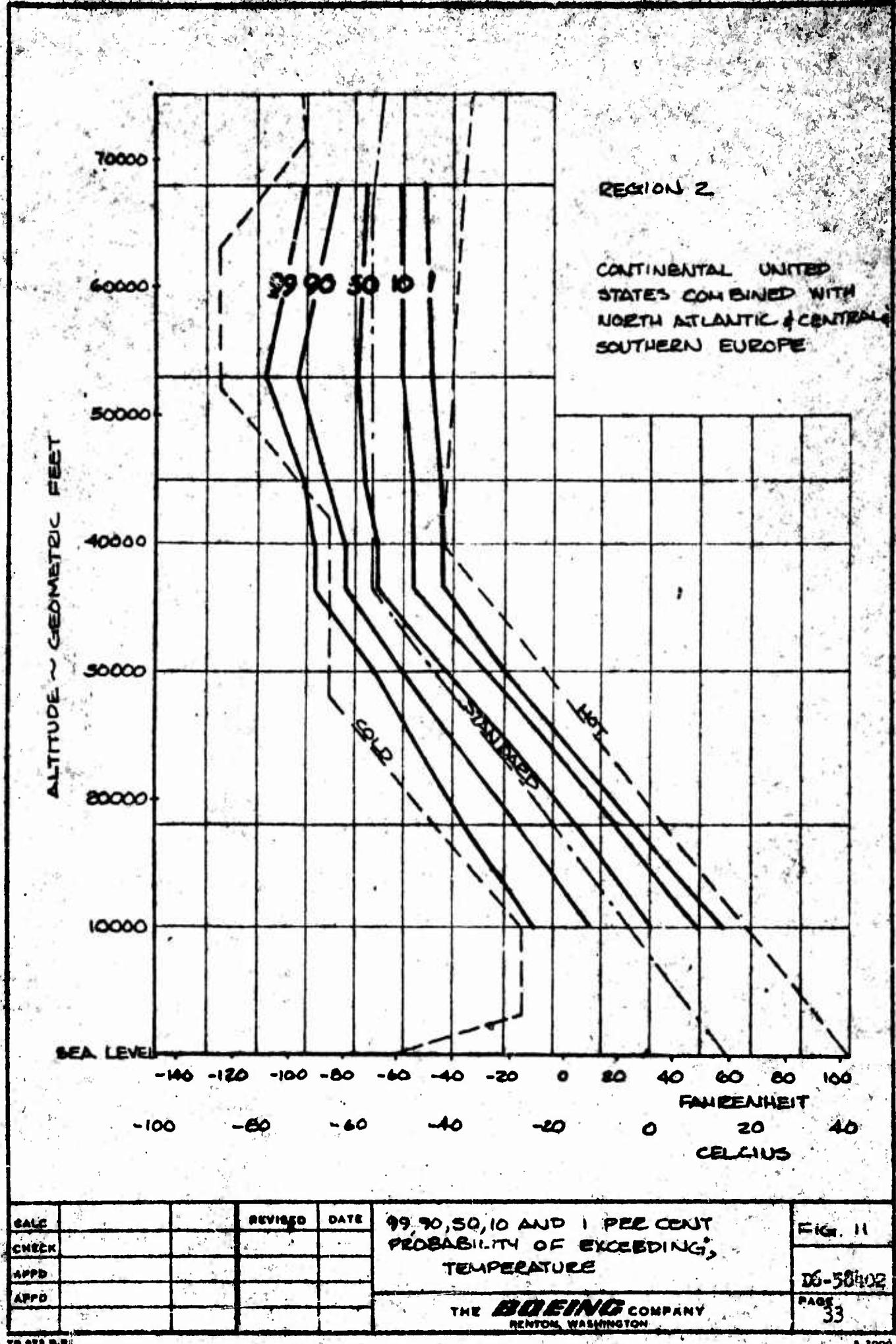
Figures 10 through 13 give the temperatures that are probable to be exceeded 99, 90, 50, 10 and 1 percent of the year over the different specified regions. The three standardized atmospheres (Standard, Hot and Cold) are superposed for comparison and reference. The temperatures at the different levels where calculations have been carried out (10,000, 18,000, 30,000, 40,000, 45,000, 53,000 and 68,000 feet) are interconnected in the most probable way.

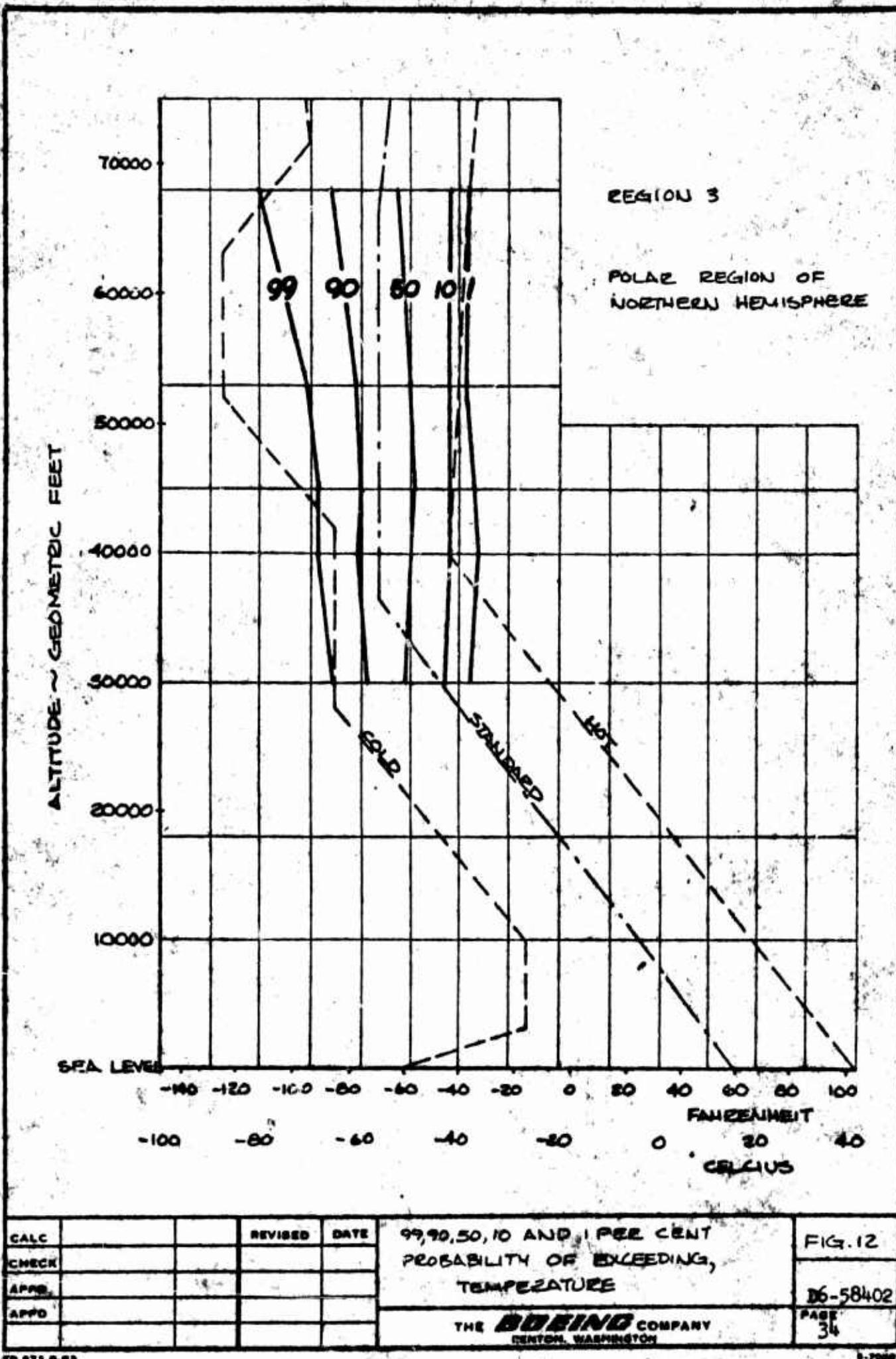
Figures 14 through 17 give the complete seasonal and annual (arithmetic mean of the four seasons) temperature distribution at 30,000 ft. altitude. Similar curves can be made available for any region and altitude. Figure 18 is a summary table of data from Figures 10 through 13, everything being referred to Standard Day condition at each altitude.

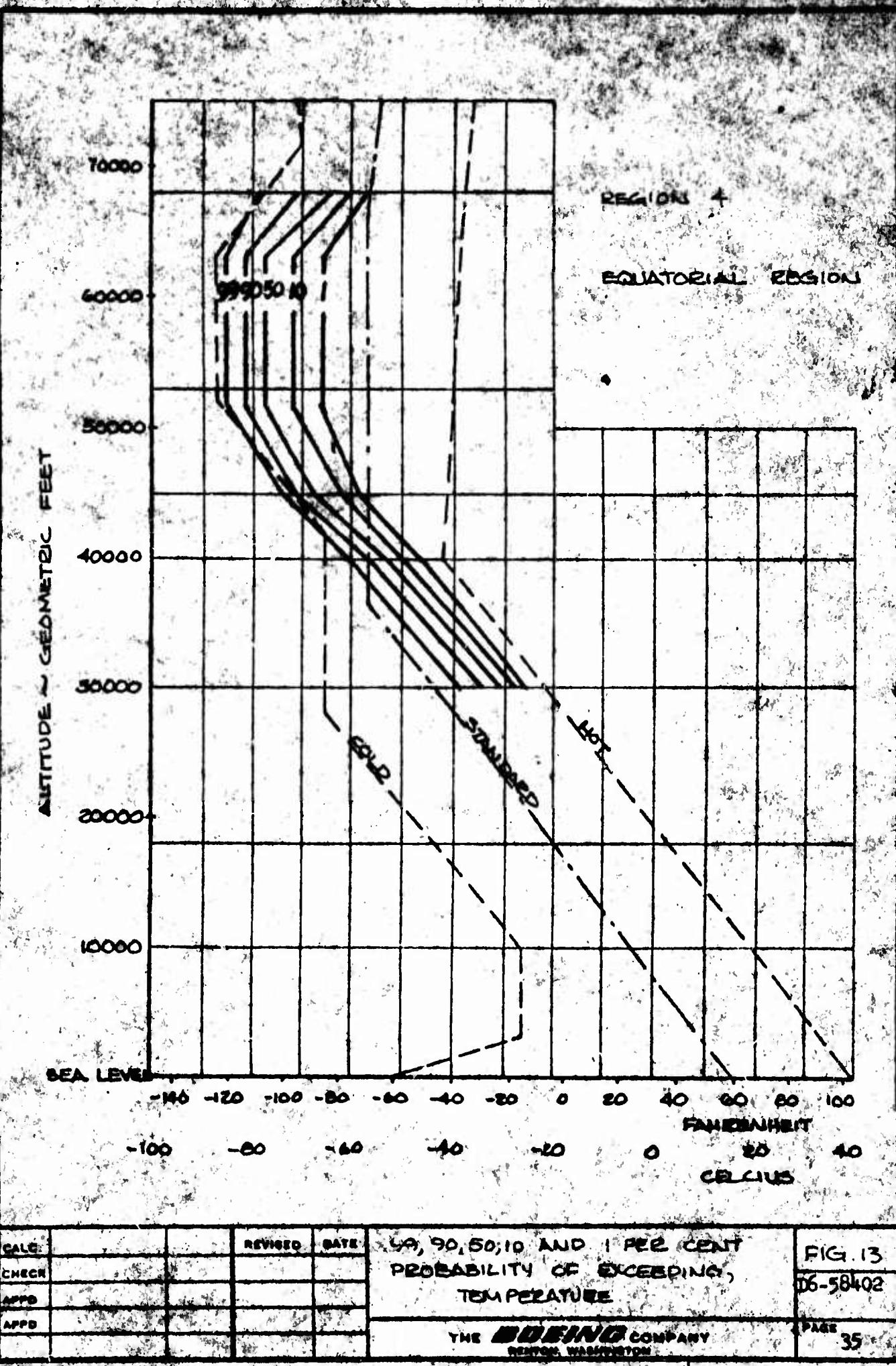


CALC		REVISED	DATE	99, 90, 50, 10 AND 1 PER CENT PROBABILITY OF EXCEEDENCE, IN TEMPERATURE	FIGURE 10
CHECK					105-500102
APPRO					
APPRO					
				THE SPERRY RAND COMPANY PRINTED IN U.S.A.	

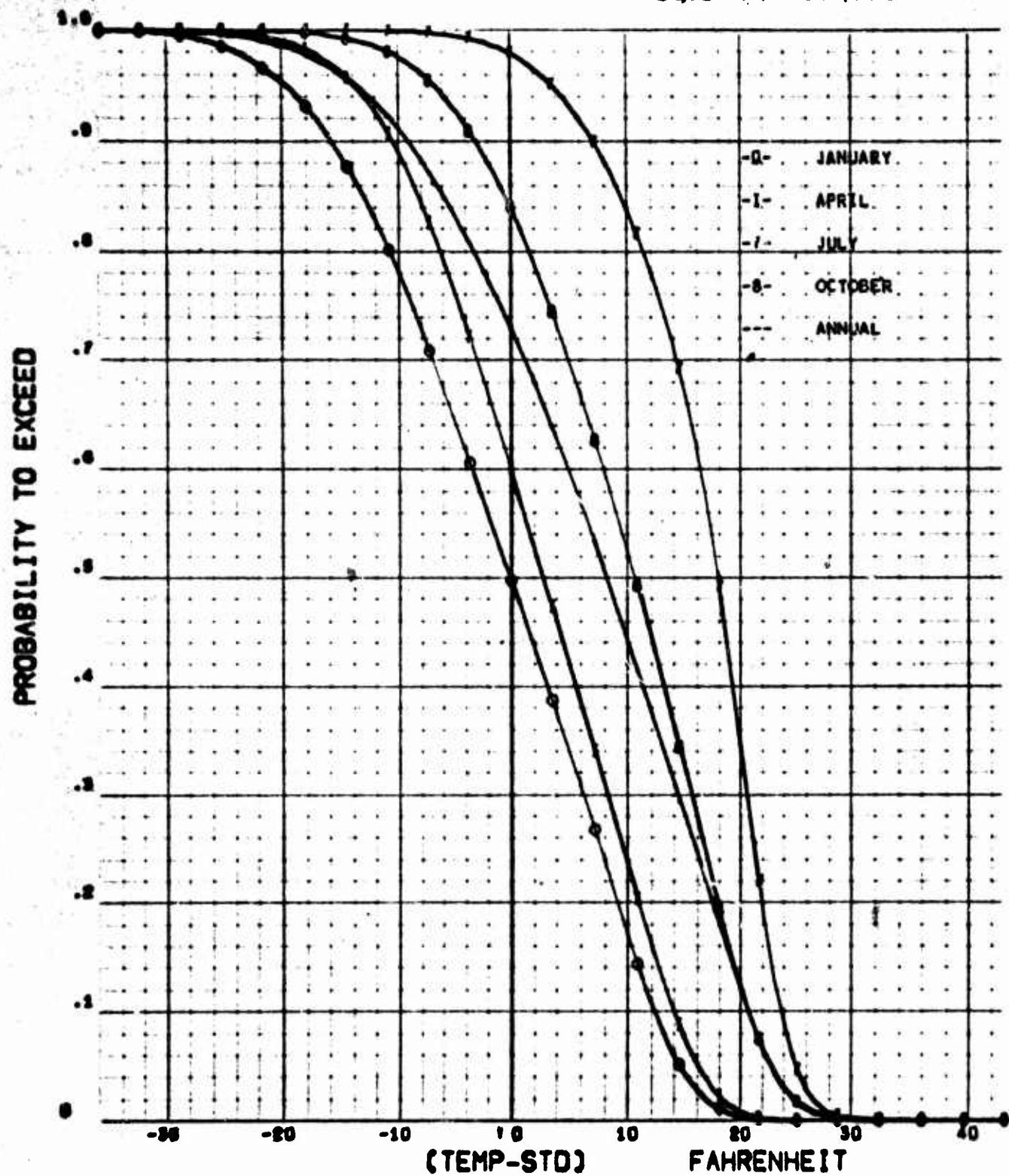
105-500102







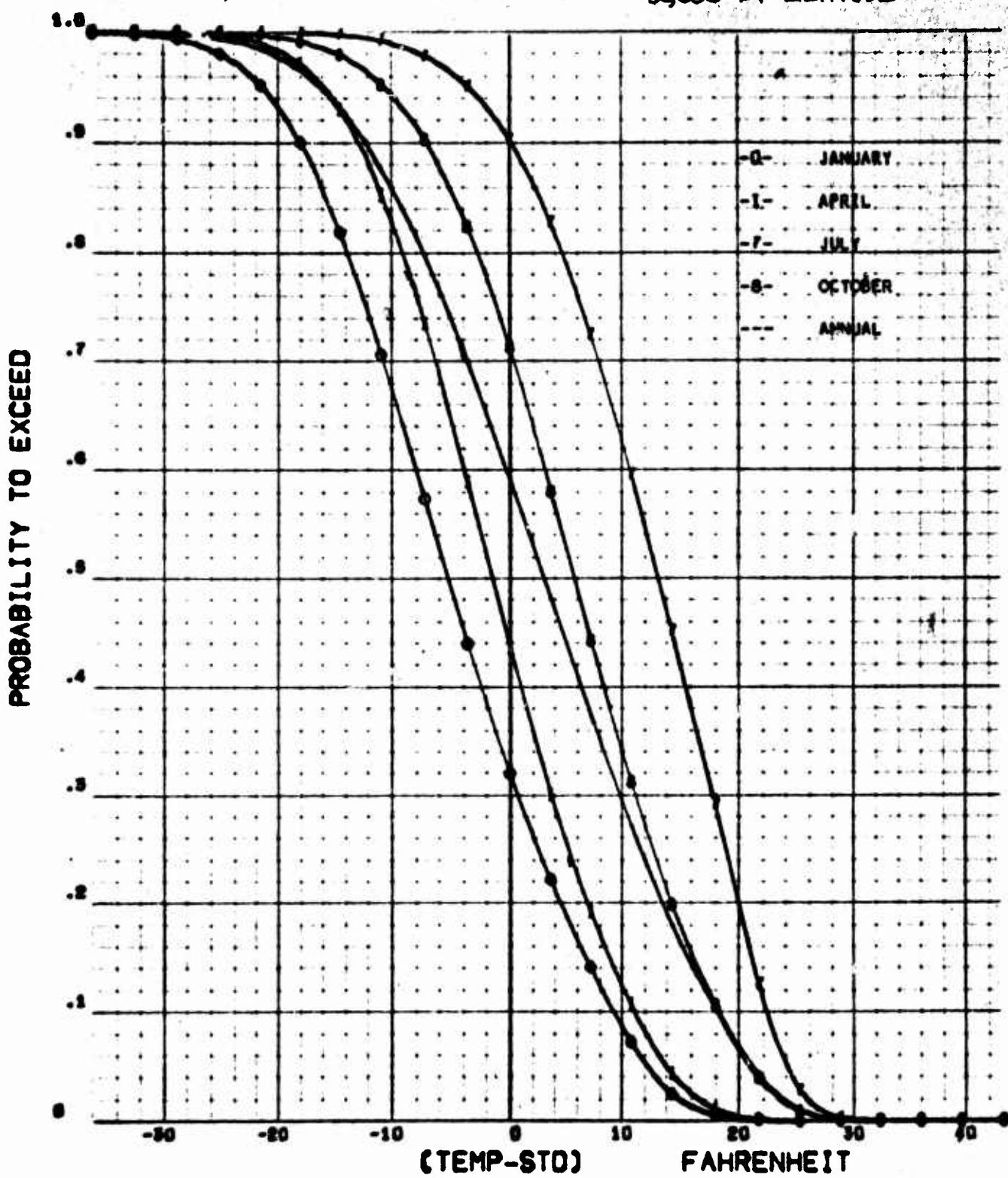
CONTINENTAL UNITED STATES
30,000 FT ALTITUDE



CALC			REVISED	DATE	SEASONAL AND ANNUAL DISTR. FOR SPEC. GEOGR. AREA	FIG. 14
CHECK						D6-58402
APPRO						PAGE
APPRO						36

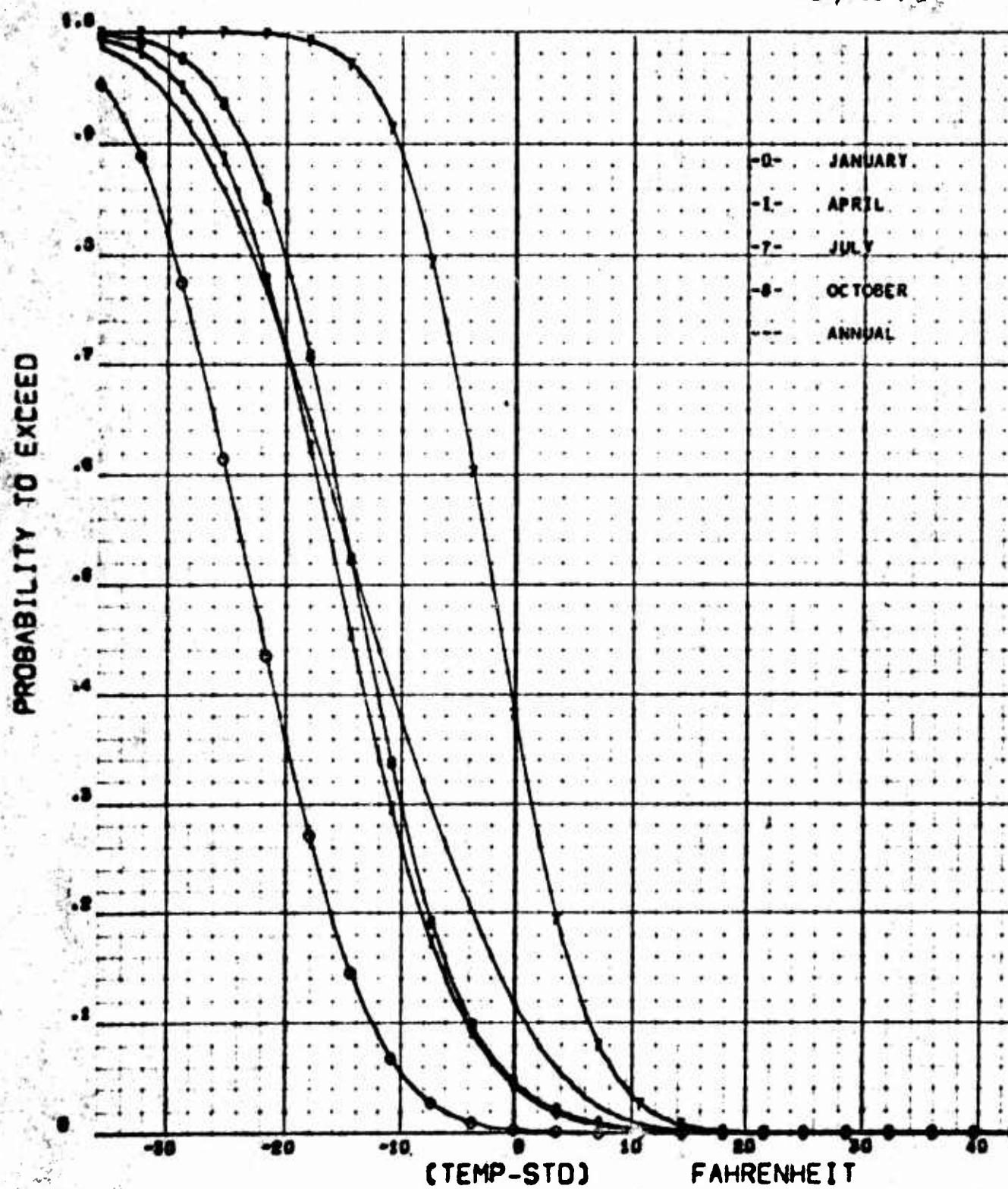
THE BOEING COMPANY
Seattle, Washington

CONTINENTAL UNITED STATES
COMBINED WITH NORTH ATLANTIC,
CENTRAL & SOUTHERN EUROPE
39000 FT ALTITUDE



CALC		REVISED	DATE	SEASONAL AND ANNUAL DISTR. FOR SPEC. GEOGR. AREA	FIG. 15
CHECK					D6-58402
APPENDIX					PAGE 37
APPENDIX				THE BOEING COMPANY INT'L. WINGMANUFACTURER	

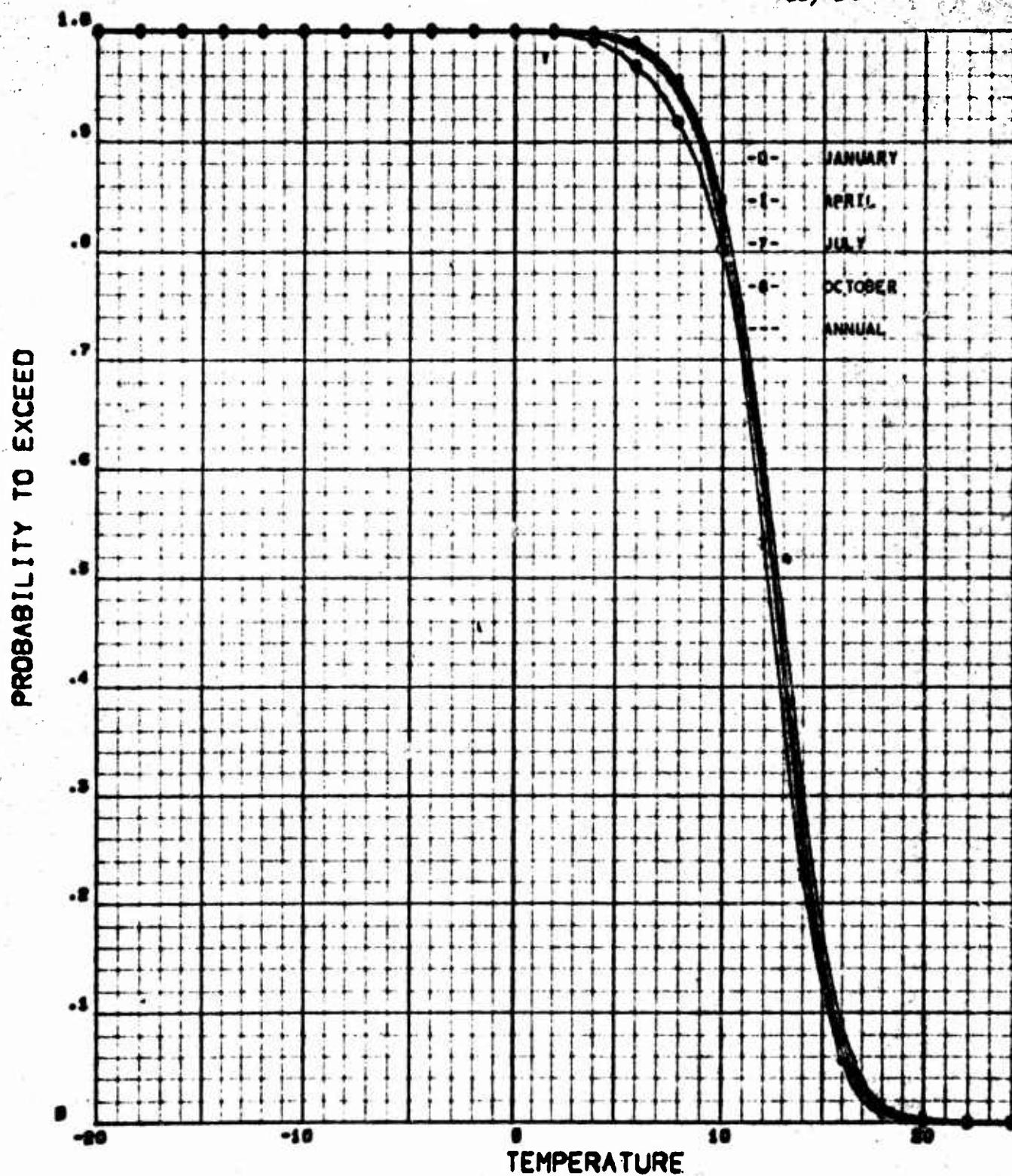
POLAR REGION
30,000 FT



CALC.		REVISED	DATE	SEASONAL AND ANNUAL DISTR. FOR SPEC. GEOGR. AREA	FIG. 16
CROSS					D6-58402
ADD					
APPD					Page 38

THE BOEING COMPANY
MANUFACTURING & ENGINEERING

EQUATORIAL REGION
30,000 FT



CALC	REVIEWED	DATE	SEASONAL AND ANNUAL DISTR. FOR SPEC. GEOGR. AREA	FIG. 17 D6-58402
ONEZ				
WESZ				
AMZ				

THE MONTGOMERY WILSON COMPANY
Montgomery Wilcox Division

39

ALTITUDE	STANDARDIZED EXTREMES		REGION	TEMPERATURE TO BE EXCEEDED ON A YEAR ROUND BASIS ~ FAHRENHEIT					
	COLD	HOT		99%	90%	50%	10%	1%	
10,000	-38	43	1	-35	-10	16	28	36	
			2	-36	-15	8	25	33	
18,000	-34	41	1	-29	-8	15	28	31	
			2	-33	-10	8	24	29	
30,000	-35	40	1	-21	-9	8	21	27	
			2	-23	-12	5	18	24	
			3	-37	-27	-14	0	9	
			4	9	16	22	28	33	
40,000	-15	25	1	-20	-9	1	11	22	
			2	-22	-10	2	11	26	
			3	-21	-7	11	26	36	
			4	-7	-1	5	12	18	
45,000	-29	27	1	-26	-20	-8	8	20	
			2	-25	-18	-2	14	25	
			3	-20	-6	13	26	84	
			4	-31	-25	-18	-9	-2	
53,000	-55	40	1	-38	-31	-16	4	16	
			2	-37	-27	-5	11	22	
			3	-25	-8	12	27	33	
			4	-53	-46	-36	-26	-14	
68,000	-37	32	1	-23	-15	-4	7	13	
			2	-25	-14	-2	10	19	
			3	-38	-21	8	26	32	
			4	-37	-25	-14	-6	0	

LEGEND: DATA APPEAR ON EACH ALTITUDE AND FOR EACH PROBABILITY LEVEL AS:

REGION 1	U.S.
REGION 2	U.S. & N.A.+E.
REGION 3	POLAR
REGION 4	EQUATOR

CALC		REVISED	DATE	TEMPERATURES REFERRED TO STANDARD ATMOSPHERE	FIG. 1B
CHECK					D6-58402
APR					PAGE 40
APR				THE BOEING COMPANY	
DRA	RUB				

PROGRAM A VOL1 INPUT, OUTPUT, TAPE5=INPUT, TAPE1, TAPE2,
1 TAPE8;

C MAIN PROGRAM, INPUT, OUTPUT

```
COMMON A,TNMP,STDN,LMS,DIST,ITHI,ITLO,MTH,PROB,TPR,REV.  
ITMA,SCALC,USSA,PRI,PIAV,TIM,NINT,DEGR,LALT,1555,EMAX,EAVR  
DIMENSION A(11,17,4,2),TNMP(57,4,2),STDN(57,4,2),DIST(57),  
IPRCG(70,5),TPR(3,5),TMN(5),SCALC(5),USSA(5),PRI(30,5),  
2PIAV(30,5),TMIN(30),DEGR(70),LALT(70),EMAX(5),EAVR(5)  
DIMENSION SEASON(4),GLAT(2),GLNG(2),TML(57,4),  
ITSOL(57,4),PRV(3),PNE(3),PROD(70),FIOD(30),UNIT(2),DEGA(70),  
INTEGER CPT(3),AL,1,2
```

$$CCF(170) = 1.8 * (170 + 40) - 40.$$

```
C ALL TEMPERATURE INPUTS IN CELCIUS (EXCEPT TINT, WHICH IS OPTIONAL)  
C INPUT NUMBER OF ALTITUDES AND HEIGHT IN FEET (IN ORDER OF  
C INCREASING ALTITUDES).
```

```
READ 130, NALT,(ALT(I)),I=1,NALT,
```

```
C INPUT OPTIONS FOR CALCULATIONS  
C OPT(1)=1 IF PRINTING OF ALL OUTPUT DATA DESIRED, =0 IF NOT  
C OPT(2)=1 IF TEMP. OUTPUT IN FAHRENHEIT, =2 IF CELCIUS  
C OPT(3)=1 IF PROBABILITIES ON TEMP. INTERVALS DESIRED, =0 IF NOT  
C TINT= SPECIFY DESIRED WIDTH OF TEMP INTERVALS (REAL). CAN BE  
OMITTED IF OPT(3)=0. CPT(2) CONTROLS UNIT OF TEMP.
```

```
READ 130,(OPT(1),IB=1,3),TINT  
IF (CPT(2)=0) EG=1, TINT=TINT/1.8
```

C PRINT HEADING

```
PRINT 150  
PRINT 160,(OPT(1),IB=1,3)
```

C MOVE REQUIRED METEOROLOGICAL DATA FROM STORAGE TAPE TO MEMORY

```
DO 200 LX=LALT  
190 READ (8) LALT(LX),USSA(LX)
```

```
READ 181
IF (LAUT(LX1) .EQ. ALTI(LX1)) 200,190
200 CONTINUE
```

```
C INPUT NUMBER OF ROUTES AND A TABLE CARD FOR EACH ROUTE
```

```
READ 130,NROUTE
DO 3000 NCASE=1,NROUTE
  READ 260
  PRINT 270,NCASE,NROUTE
  PRINT 260
  PRINT 275
```

```
C INPUT GEOGRAPHICAL LOCATION OF START AND END POINT OF ROUTE
```

```
READ 290, (GLAT(I),GLNG(I),I=1,2)
155S=0
CALL TEMP(GLAT,GLNG,NALT)
```

```
C TEMP FURNISHES TEMP,MEAN AND STDY FOR POINTS ABOUT ZONN. APART  
C ALONG THE ROUTE IN ALL SEASONS AND ALTITUDES
```

```
DO 2000 LVL=1,NALT
```

```
15 (155S,NE,0) GO TO 2000
IF (LVL.EQ.1) 1360,1390
1360 PRINT 1280, (GLAT(I),GLNG(I),I=1,2)
1390 PRINT 1400, LAT(LVL)
```

```
C FIND TEMP RANGE TO BE USED IN CALC. ON EACH PARTICULAR ALTITUDE  
C TRANSFORM MEAN AND STDY INTO 2-DIM. ARRAYS
```

```
TMIN=TNMP(1,1,LVL)
TMAX=TMIN
DO 360 IB=1,4
DO 350 IA=1,L42
  IF (TNMP(IA,IB,LVL).LT.TMIN) TMIN=TNMP(IA,IB,LVL)
  IF (TNMP(IA,IB,LVL).GT.TMAX) TMAX=TNMP(IA,IB,LVL)
  TM(IA,IB)=TNMP(IA,IB,LVL)
  TSDL(IA,IB)=STDY(IA,IB,LVL)
350 CONTINUE
360 CONTINUE
```

```

1100= IFIX(JMIN)-15
11TH= IFIX(JMAX)+15
DO 1000 MTH=1,4
  CALL DISTR(MTH, TSDL, LM3)

C DISTR GIVES THE PROB. TO EXCEED THE DIFFERENT TEMP. VALUES
C OF THE ABOVE DEFINED TEMP. RANGE

C FIND THE TEMP. FOR WHICH THE PROB. NOT TO EXCEED IS .50+.75+.85

DATA (PRVIA),IA=1,3/.50,.25,.15/
DO 500 IA=1,3
  PRVIA= 1.+PRVIA
JX=1
420 IF (PFDG(JX,MTH).GT.PRV(IA)) 430,480
430 JX=JX+1
  GO TO 420
480 TPRIA,MTH)= DEGR(JX)-(PRVIA)-(PROB(JX,MTH)/(PROB(JX-1,MTH)-
500 CONTINUE
  CALL FITNC
C FITNC FINDS A NORMAL DISTR. CURVE WHICH FOLLOWS THE UNIQUE DISTR.
C CURVE AS CLOSELY AS POSSIBLE
IF (OPT(3)=EQ.0) GO TO 1000
  CALL PINTINT(LVL)
C PRINT GIVES PROB. ON TEMP. INTERVALS ABOUT THE USSA-TEMPERATURE
1000 CONTINUE
C CALCULATE THE ANNUAL CASE BASED ON DATA FOR THE FOUR SEASONS.
C ANNUAL DATA ARE STORED IN COLUMN FIVE (MTHS) OF EXISTING
C ARRAYS FOR SEASONAL DATA
MTH= 5
DO 1100 JA=1,ITR
  PSAS= 0.
  DO 1080 IS=1,4

```

```

1080 PSA= PSA+PROB(JA,IS)
1100 PROB(JA,51)= .25*PSA
DO 1140 IB=1,3
JX=1
1126 IF (PROB(JX,51).GT.PROB(JX,5)) JX=1140
1125 JX= JX+1
GO TO 1120
1140 IPRI(1,5)= DEGR(JX)-(PRV(1B)-PROB(JX,5))/1
1141 (PROB(JX-1,5)-PROB(JX,5))

      CALL FITNC

      IF (CRT(3)*EG.0) GO TO 1190
      CALL PINT(TINT,LVL)
1190 IF (OPT(2)*NE.1) GO TO 1300

C CONVERT TEMP.-ARRAYS FROM CELCIUS TO FAHRENHEIT

      DO 1200 J=1,ITK
1200 DEGR(J)= CCF(DEGR(J))
      DO 1220 IB=1,5
      DO 1220 IA=1,3
1220 TPRI(IA,IB)= CCF(IPRIA,IB))
      DO 1230 K= 1,NINT
1230 TIM(K)= 1.8*TIM(K)
      DO 1235 M=1,5
1235 TMIN(M)= CCF(TMIN(M))
      1235 SCALC(M)= 1.8*SCALC(M)
1300 CONTINUE

C PRINT CALC. SEASONAL AND ANNUAL DATA FOR THE PARTICULAR ALTITUDE

      PRINT 1410,(SEASON(N),N,S=1,4)
      IF (REV*EG.0) 1412,1414
1412 PRINT 1415, (DIST(I)*(TMAP(LM3+1-I*MTH*LVL)),STD(LM3+1*MTH*LVL))
      MTH=1,4,I=1,LM3
      GO TO 1416
1414 PRINT 1415, (DIST(I)*(TMAP(LM3+1-I*MTH*LVL)),STD(LM3+1*MTH*LVL))
      MTH=LVL,I=1,4,MTH=1,LM3
1418 IT= OPT(2)
      PRINT 1420, UNIT(IT)

```

DO-58402

Page 44

```

IF (OPT(1)=EQ•J) GO TO 1450
PRINT 1430, (SEASON(NS),NS=1•4)
PRINT 1435, DEGR(J)•(PROB(J,MTH)•MTH=1•5),J=1•11R
1450 PRINT 1460, (SEASON(NS),NS=1•4)
PRINT 1465, (PNE(K),TPK(K,MTH),MTH=1•5),K=1•31
IF (OPT(1)=EQ•CR•OPT(3)=EQ•0) GO TO 1500
PRINT 1480, (SEASON(NS),NS=1•4)
PRINT 1495, (TM(1),PINVL,MTH),MTH=1•5),L=1•NINT
PRINT 1510, (SEASON(NS),NS=1•4)
PRINT 1515, (TM(MTH),MTH=1•5),(SCALC(MTH),MTH=1•5)
PRINT 1530, (EAVR(MTH),MTH=1•5),(EMAX(MTH),MTH=1•5)

```

C WRITE GRAPHICAL OUTPUT DATA ON BINARY TEJD FOR PROCESSING BY
C THE SC-4020 MACHINE

```

IF (INCASE•AND•LVL•EW•1) 1610,1615
1610 IRR= 0
1615 IF (IRR•EG•0) 1630,1620
1620 PRINT 1625, IRR
      GO TO 2000
C ADJUST TEMP•RANGE TO ATTAIN OPTIMUM READABILITY ON PLOTS
1630 I1= 0
      I2= 0
      DO 1650 J=1•11R
      IF (I2•EQ•0.5) GO TO 1650
      I2= 0
      DO 1640 MTH=1•5
      IF (PROB(J,MTH)•LT•0.995•AND•I1•EQ•0) 1635,1638
      I1= 1
      1638 IF (PROB(J,MTH)•LT•0.005) I2= 12+1
      1640 CONTINUE
      JCH= J
      1650 CONTINUE
      ITC= JCH-JCL+1
      DO 1655 J= JCL,JCH
      JC= J-JCL+1
      1655 DEGA(JC),DEGR(J)
      DO 1720 MTH= 1•5
      NT APE= 1
      DO 1660 J= JCL,JCH

```

JC=J-JEL+1
1660 PROB(JC)=PROB(J,MTH)
CALL WBIN4DEGA,PROD4JL(CTAPE,IRRI)

C WBIN FEEDS ARGUMENTS TO THE 6600-SYSTEM TAPE WRITING SUB-
ROUTINE WRITP

IF(IOPT(3)=EQ.0) GO TO 2000
NTAPE=2
DO 1680 K=1,NINT
PICD(K)=PINV(K,MTH)
1680 CALL WBIN(CTIM,PIOD,NINT,NTAPE,IRRI)
2000 CONTINUE
3000 CONTINUE

DATA (SEASON(NS),NS=1,4)/7HJANUARY,3HAPRIL,4MJULY,7HOCTOBER/
DATA (UNIT(IT),IT=1,2)/10HFAHRENHEIT,10H CELCIUS /
130 FORMAT (3I6,F12.2)
150 FORMAT (1//,20X,*SEASONAL AND ANNUAL TEMP. DATA FOR GREAT
1CIRCLE AIR ROUTES*)
160 FORMAT (1//20X,*OPT(1)=*,13.5X,*OPT(2)=*,13.5X,*OPT(3)=*,13.)
260 FORMAT (80H
)
270 FORMAT (*1*,20X,*CASE NO.*,I2,* OUT OF NSOLTE=*,15//)
275 FORMAT (10X,45H*****
290 FORMAT (4F12.2)
1580 FORMAT (1/10X,*START POINT LATITUDE =*,F8.2/22X,*LONGITUDE =*,
1F8.2/10X,*END POINT LATITUDE =*,F8.2/22X,*LONGITUDE =*,F8.2)
1400 FORMAT (1//10X,*ALITUDE=*,17.7*FT*/10X*20H*****
1410 FORMAT (1//10X,*MEAN ST. DEV. FOR POINTS ALONG THE ROUTE*)
15X,15H** CELCIUS ***/10X,*DIST NO.=*,3X*4(1A9.7X)/
1415 FORMAT (12X,F4.0,2X*4(7X*FS.4*/*F3.0))
1420 FORMAT (1//10X,* TEMPERATURE UNIT IN FOLLOWING TABLES :5*,
15X,4H** A10.4H ***)
1430 FORMAT (1//10X,*PROBABILITY TO EXCEED GIVEN TEMP. VALUES*/10X,
1*TEMP.,11X,4(1A9.3X)*ANNUAL*)
1435 FORMAT (10X,F6.1*3X*5(5X*F7.5))
1460 FORMAT 4//10X,*TEMP. THAT CORRESPOND TO GIVEN PROB. N C T RRP.
1* BE EXCEEDED */10X,*PROB. VALUE *,5X*4(A9.3X),*ANNUAL*)
1465 FORMAT (14X*F4.2,5(7X*F5.1))
1480 FORMAT (1//10X,*PROBABILITY ON INTERVALS ABOUT THE USSA-TEMP*)

1/10X,*INT*MEDIAN*,6X,4(A9,3X)*ANNUAL*/)
1485 FORMAT (12X,*STD*,F5.1*5(5X,F7.5))
154C FORMAT (//1CA*FIT A NORMAL CURVE TO THE*)
1* ROUTE DISTR,* /25X*4(A9,3X)*ANNUAL*/)
1515 FORMAT (10X,*MEAN TEMP*,5(5X,F7.2)/10X,
1*ST. DEV. =*,5(5X,F7.2))
1530 FORMAT (//10X,*RESEMBLANCE OF FITTED CURVE TO ROUTE DISTR*/
110X,*DEVIATION IN PROBABILITY VALUES FOR CALC. TEMPERATURES*/
210X,*ARIT.MEDIAN DEV.=*,5(F5.3,7X)/10X,*MAXIMUM DEVO. **,
35(F5.3,7X))
1625 FORMAT (///*AN ERROR SENSED IN WRITING ON BINARY TEJPS/
1*SEE DCC. D6-19175TH FOR ANALYSIS - ERROR KEY) IRR=**14/
2*ALL FURTHER WRITING ON PLOT-PROCESS TEJPS BYPASSED*)
ENDFILE 1
ENDFILE 2
END

SUBROUTINE TEMP(GLAT,GLNG,NALT)

C TEMP CALCULATES MEAN- AND ST. DEV. DATA FOR POINTS ABOUT 260N.M.
C APART ALONG THE GIVEN ROUTE.

```

COMMON A,TMP,STDN,LM3,DIST,ITHI,ILO,ITR,MH,PROB,TPR,REV,
1 TMN,SCALC,USSA,PIN,PRN,TIM,NINT,DEGR,LALT,ISSS,EAX,EAVR
2 DIMENSION A(1117,4,2),TMP(3,7,4,2),STDN(57,4,2),DIST(57),
3 PROB(70,5),PR(3,5),TMN(5),SCALC(5),USSA(5),PR(30,5),
4 PIN(30,5),TIM(30),DEGR(70),LALT(5),EAX(5),EAVR(5)
5 DIMENSION CPHI(2),CLMB(2),TOPHI(2),TOLMB(2),TRPHI(2),
6 TRLMB(2),RPHI(57),RLMB(57),GLAT(2),GLNG(2)

```

C SPECIFY DATA DECODING AND CONVERT ROUTE COORDINATES

```

TEMPF(X)=FLOAT(IFIX((X/100.))/10.
STDF(X)=ABS((X-TEMPF(X)*1000.)/100.)
PTNF(W,X,Y,Z)=W*FLOAT(IFIX(PHI/5.))+X*FLOAT(IFIX(AMB/Y))+Z
IRNDF(X)=X+SIGN(.5,X)
RADEG=1.74533E-02
DEGRA=57.29578
RE = 3.4377468E3

```

```

3003 DO 2009 I=1,2
CPHI(1)=(GLAT(1)-FLOAT(IFIX(GLAT(1)))*5./3.+FLOAT(IFIX(GLAT(1))))
2009 CLMB(1)=(GLNG(1)-FLOAT(IFIX(GLNG(1)))*5./3.+FLOAT(IFIX(GLNG(1)))
DO 21 I=1,2
TOPHI(1)=90.-CPHI(1)
IF (CLMB(1)) 22,22,23
22 TOLMB(1)=360.+CLMB(1)
GO TO 24
23 TDLMB(1)=CLMB(1)
24 TRPHI(1)=TOPHI(1)*RADEG
21 TRLMB(1)=TDLMB(1)*RADEG
REV=0.
IF (ABS(TDLMB(1)-TDLMB(2))-180.151,51,52
SAVE=AMAX1(TRLMB(1),TRLMB(2))
TRLMB(2)=AMIN1(TRLMB(1),TRLMB(2))
TRLMB(1)=SAVE
GO TO 53
52 SAVE=AMIN1(TRLMS(1),TRLMS(2))

```

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```

      TRLMB(2)=AMAX1(TRLMB(1),TRLMB(2))
      TRLMB(1)=SAVE
      TDLMB=TRLMB(2)/RADEG
      IF(ABS(TDLMB1-TDLMB2)-0.001) 54,55,55
  53  SAVE=TRPHI(2)
  54  TRPHI(2)=TRPHI(1)
      TRPHI(1)=SAVE
      REV=1.

```

54 CALCULATE POINTS ON THE GREAT CIRCLE ROUTE

```

  55  COSPH1=COS(TRPHI(1))
      COSPH2=COS(TRPHI(2))
      SINPH1=SIN(TRPHI(1))
      SINPH2=SIN(TRPHI(2))
      TRDLNG=TRLMB(2)-TRLMO(1)

  56  GCUA=ACOS(COSPH1*COSPH2+SINPH1*SINPH2*COS(S(TRLNGL)))
      GCD=RE*ABS(GCDA)
      IGGD = GCD+0.5
  57  INT=AMAX1(GCD/200.0*1.1)
      GCSGRA=GCDA/FLOAT(INT+1)
      GCDA=ABS(GCDA)
      SIND=SIN(GCDA)
      ANGLE=2.*ASIN(GSRT((COS(GCDA-TRPHI(1))-COSPH2)/2.+SINPH1/SIND))

  58  RPHI(1)=TRPHI(1)
      RPHI(1+2)=TRPHI(2)
      RLMB(1)=TRLMB(1)
      RLMB(1+2)=TRLMB(2)
      LM2=INT+1
      DO 47 IPT=2,LM2
      SIND=SIN(GGGRA*FLOAT(IPT-1))
      RPHI(IPT)=ACOS((COS(GCSGRA+FLOAT(IPT-1))*COSPH1+SIND)*SINPH1*COS
      RLMB(IPT)=TRLMB(1)-2.*ASIN(SUR((COS(TRPHI(1)-RPHI(IPT))-COS
      RPHI(IPT))=ACOS((COS(GCSGRA+FLOAT(IPT-1))/2.*SINPH1/SIN(RPHI(IPT))))*
      1.17857
      IF(TRLMB(IPT)) 41,47,47
  41  RLMB(IPT)=283185+RLMB(IPT)
  47 CONTINUE

```

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INTERPOLATION AT ROUTE POINTS

```

LM3=LM2+1
DO 2100 IPT=1,LM3
    PHI=FLOAT(IFIX(IPHI*(IPT)*100.*DEGRAY)/100.
    AMB=FLOAT(IFIX((RLMB*(IPT))*50.*DEGRA)/50.)
    PHIMOD=AMOD(IPHI,5.)
    CF1=PHIMOD
    IF (PHIMOD) 701,702,701
    701 IF (AMOD(AMB,20.)) 706,1700,706
    702 IF (AMOD(AMB,20.)) 703,1800,703
    703 IF (AMOD(AMB,10.)) 1600,704,1600
    704 IF (IPHI-150.) 705,1800,1600
    705 IF (IPHI-30.) 1600,1800,1800
    706 IF (AMOD(AMB,10.)) 707,708,707
    707 IF (IPHI-5.) 1200,2000,709
    708 IF (IPHI-150.) 711,2000,713
    709 IF (IPHI-175.) 1100,2000,1500
    711 IF (IPHI-30.) 712,2000,1700
    712 IF (IPHI-25.) 714,2000,1300
    713 IF (IPHI-155.) 1400,2000,715
    714 IF (IPHI-5.) 1400,2000,1100
    715 IF (IPHI-175.) 1100,2000,1300

```

C THE GENERAL 4-POINT CASE

```

1100 IF (IPHI-25.) 1101,2000,1102
1101 C=20.
D=3.
H=-17.
GO TO 1103
1102 IF (IPHI-30.) 1104,2000,1105
1104 C=10.
D=98.
G=20.
H=76.
GO TO 1106
1105 IF (IPHI-150.) 1107,2000,1108
1107 C=10.
D=-87.
H=-125.

```

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```

A=37.
60 TO 1109
1108 IF (PHI-155.) 1110,2000,1111
1110 C=20.
D=1023.
G=10.
H=985.
GO TO 1106
1111 C=20.
D=453.
H=433.
1103 A=19.
1109 G=C
GO TO 1112
1106 A=0.
1112 N1=PTNF(A,100CD)
N3=PTNF(A,100G,H)
CF2=AMOD(AMB,C1)
CF3=AMOD(AMB,C1)
DO 1113 LVL=J+NALT
DO 1113 MTH=1,4
TNMP(IPI,MTH,LVL)=(PHIMOD*(TMEPF(A(N1+1)*
XTH*LVL)*(C-CF2))/C+(5.-PHIMOD)*(TEMPF(A(N3+1)*
XF(A(N3+1)*MTH*LVL))*CF3)/G)/5.
1113 STDN(IPI,MTH,LVL)=(PHIMOD*(STDVF(A(N1+1)*
XTH*LVL)*(C-CF2))/C+(5.-PHIMOD)*(STDVF(A(N3+1)*MTH*LVL))*CF3)/G)/5.
GO TO 2100

```

C THE GENERAL 3-PCINT CASE WITHIN 5 DEGREES OF THE NORTH POLE 1200

```

1200 N1=PTNF(0.,0.,1.,20.,0.,3.)
N3=PTNF(0.,0.,1.,1.)
CF1=AMOD(AMB,20.)
DO 1201 LVL=1,NALT
DO 1202 MTH=1,4
TNMP(IPI,MTH,LVL)=(PHI*(CF1*TEMPE(A(N1+1)*
X(N1-1)*MTH*LVL))/20.+(5.-PHI)*(TEMPF(A(N3+1)*
CF1*STDVF(A(N1+1)*MTH*LVL))/20.-CF1)*STDVF(A(N3+1)*
STDVF(A(N1+1)*MTH*LVL))/5.
GO TO 2100

```

THE SPECIAL 3-POINT CASE WITHIN 5 DEGREES OF THE SOUTH POLE CR 1300
THE SPECIAL 3-POINT CASE BETWEEN 25 AND 30 DEGREES LATITUDE

1300 IF (PHI-30.0) 1301,2000,1302

1301 D=79.

F=16

H=97.

GO TO 1303

1302 D=1099.

F=0.

H=1117.

1303 N1=PTNF(0.,0.,1.,0.,20.,0,D)
N3=PTNF(0.,0.,F,10.,0,H)
DO 1304 LVL=1,MALT
DO 1304 MTH=1,4
TNMP(IPT,MTH,LVL)=(5.-PHIMOD)*(TEMPF(A(N1,MIH,LVL))+TEMPF(A(N1-1,0,
XTH,LVL))/10.+PHIMOD*TEMPF(A(N3,MIH,LVL))/2.
1304 STDN(IPT,MIH,LVL)=(5.-PHIMOD)*(STDVF(A(N1,MIH,LVL))+STDVF(A(N1-1,0,
XTH,LVL))/10.+PHIMOD*STDVF(A(N3,MIH,LVL))/5.
GO TC 2100.

THE SPECIAL 3-POINT CASE WITHIN 5 DEGREES OF THE NORTH POLE CR 1400
THE SPECIAL 3-POINT CASE BETWEEN 150 AND 155 DEGREES LATITUDE

1400 IF (PHI-5.0) 1401,2000,1402

1401 D=3.

F=0.

H=1.

GO TO 1403

1402 D=1023.

F=1.

H=985.

1403 N1=PTNF(0.,0.,1.,0.,20.,0,D)
N3=PTNF(0.,0.,F,10.,0,H)
DO 1404 LVL=1,MALT
DO 1404 MTH=1,4
TNMP(IPT,MIH,LVL)=PHIMOD*(TEMPF(A(N1,MIH,LVL))+TEMPF(A(N1-1,0,0,
XL))/10.+((5.-PHIMOD)*TEMPF(A(N3,MIH,LVL))/5.
1404 STDN(IPT,MIH,LVL)=PHIMOD*(STDVF(A(N1,MIH,LVL))+STDVF(A(N1-1,0,
XL))/10.+((5.-PHIMOD)*STDVF(A(N3,MIH,LVL))/5.

60 10 2100

C THE GENERAL 3-POINT CASE WITHIN 5 DEGREES OF THE SOUTH POLE 1500

```
1500 N1=PTNF(C,10.20,.1099.)
N3=PTNF(0,0,1.11175)
CF1=AMOD(C,ANB,20.)
DO 1501 LVL=1,NALT
DO 1501 MTH=1,4
TNMP(IPT,MTH,LVL)=(180.-PHI)*(CF1*TEMPF(A(N1,MTH,LVL))+(20.-CF1)*
XEMPF(A(N1-1,MTH,LVL))/100.+((PHI-175.)*TEMPF(A(N3,MTH,LVL))/15.
1501 STDNF(IPT,MTH,LVL)=(180.-PHI)*(CF1*STDVF(A(N1,MTH,LVL))+(20.-CF1)*
XDVVF(A(N1-1,MTH,LVL))/100.+((PHI-175.)*STDVF(A(N3,MTH,LVL))/15.
50-TC 2100
```

C THE SPECIAL 2-POINT CASE ON A LATITUDE AN EVEN MULTIPLE OF 5 1500

```
1600 IF (PHI-30.) 1601,1602,1602
1601 D=-16.
      GO TO 1603
1602 IF (PHI-150.) 1604,1604,1605
1605 D=434.
1603 A=19.
C=20.
      GO TO 1606
1604 A=37.
C=10.
D=-124.
      GO TO 1606
1606 N1=PTNF(A,10.,C,DI)
CF1=AMOD(C,ANB,C)
DO 1607 LVL=1,NALT
DO 1607 MTH=1,4
TNMP(IPT,MTH,LVL)=(TEMPF(A(N2,MTH,LVL))+CF1+15MPF(A(N1-1,MTH,LVL))
X*(C-CF1))/C
1607 STDNF(IPT,MTH,LVL)=(STDVF(A(N2,MTH,LVL))+CF1+STDVF(A(N1-1,MTH,LVL))
X*(C-CF1))/C
      GO TO 2100
```

C THE SPECIAL 2-POINT CASE ON A LATITUDE AN EVEN MULTIPLE OF 20
OR AN EVEN MULTIPLE OF 10 BETWEEN -30 AND +90 DEGREES LATITUDE 1700

1700 IF (PHI=5.) 1701,20000,1702
1701 A=0.
B=1.
C=20.
D=2.
F=0.
G=1.
H=1.

GO TO 1702
1702 IF (PHI-25.)* 1704,20000,1705
1704 D=2.
G=20.
H=-17.

GO TO 1706
1705 IF (PHI-30.)* 1707,20000,1708
1707 A=0.
C=10.
D=97.
G=26.
H=78.

GO TO 1709
1708 IF (PHI-150.)* 1710,20000,1711
1710 A=37.
C=10.
D=-88.
G=10.
H=-125.

GO TO 1709
1711 IF (PHI-155.)* 1712,20000,1713
1712 A=0.
D=1022.
G=10.
H=985.

GO TO 1714
1713 IF (PHI-175.)* 1715,20000,1716
1715 D=452.
G=20.
H=433.
1706 A=10.
1714 C=20.
1709 H=1.

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F=1.
GO TO 1703

1716 A=0.

B=0.

C=1.

D=11117.

E=1.

F=1.

G=20.

H=1098.

1703 E=A

N1=PTNFI(A,B,C,D)

N3=PTNFI(E,F,G,H)

DO 1717 LVL=1,NALT

DO 1717 MTH=1,4

TRIM(PIPT,MTH,LVL)=(TEMPF(A(N1,MTH,LVL))*PHINOD+TMDF(A(N2,MTH,LV

X1)*((5.-PHINOD))/5.

1717 SINDC(PII,MTH,LVL)=(STDVF(A(N1,MTH,LVL))*PHINOD+STUVF(A(N3,MTH,LV

X1)*((5.-PHINOD))/5.

GO TO 2100

C THE SPECIAL 1-POINT CASE ON A DATA POINT

1800 IF (PHI) 2000,1601,1602

1801 D=1.

GO TO 1803

1802 IF (Phi-30.) 1804,1805,1805

1804 D=-17.

GO TO 1806

1805 IF (Phi-180.) 1808,1807,2000

1807 D=1117.

1803 A=0.

B=0.

GO TO 1809

1809 IF (Phi-150.) 1811,1811,1810

1810 D=433.

1806 Z=1.

A=19.

1809 C=20.

GO TO 1812

1811 D=-125.

A=37.

```
B=1  
C=103  
1812 NI=PTNFIA,B,C,D  
DO 1813 LVL=1,NALT  
DO 1813 MTH=1,4  
    INMP(IPT,MTH,LVL)=TEMPP(AIN1,MTH,LVL)  
    1813 STDN(IPT,MTH,LVL)=STDWF(AIN1,MTH,LVL)  
  
2100 CONTINUE
```

```
C ACCUMULATED FLIGHT DISTANCE
```

```
F1=FLOAT((IGCU)/FLOAT(LM2))  
F2=0.C  
D1 450 IPT=1,LM3  
DIST(IPT)= FLOAT(IRNDF(F2))  
F2=F2+F1  
450 CONTINUE
```

```
C CHECK FOR GEOGRAPHIC REGION WHERE NO RAW DATA AVAILABLE
```

```
SLD= 3.1415927*29./36.  
DO 2500 LVL=1,NALT  
IF (LALT(LVL)*EG*68000) 2400,2500  
2400 DO 2500 IPT=1,LM3  
IF IRPHI(IPT).GT.SLD) 2420,2500  
2420 LSS= 1  
PRINT 2460  
2460 FORMAT(*METEOROLOGICAL DATA NOT AVAILABLE ON 68000FT*,  
*1* SOUTH OF 555. LATITUDE//SPECIFIED ROUTE ENTERS*,  
*2* THIS REGION*)  
2500 CONTINUE  
GO TO 3000  
2000 WRITE OUTPUT TAPE 6,2001  
2001 FORMAT(1H0$X,9HAN ERROR OCCURRED IN INTEGER ARITHMETIC)  
3000 RETURN  
END
```

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SUBROUTINE DIST(XMEAN,XSTD,NP)

C CALCULATES THE PROB. TO EXCEED THE DIFFERENT TEMP. VALUES
C OF THE ADOPTED TEMP. RANGE

```
COMMON A,TNAP,SIDN,LNG,DIST,IITH,IITL0,IITR,MTH,PROB,TPR,REV,  
IJMN,SCALC,USSA,PRI,PIV,TIM,NINT,DEGR,LAL1,I555,EMAX,EAVR  
DIMENSION A(1117,4,2),TNP(57,4,2),STOND(7,4,2),DIST(57)  
IPACB(70,5),TPR(3,5),TMN(5),SCALC(5),USSA(5),PRI(30,5),  
2PIV(30,5),TIM(30),DEGR(70),LALT(5),EMAX(5),EAVR(5)  
DIMENSION XMEAN(57,4),XSTD(57,4)  
ITR=IITH-IITL0+1  
DO 80 J=1,ITR  
PSUM=0.  
DEGR(J)=IITL0+J-1  
DO 70 I=1,NP  
PIVD=1.0-CNORMAL(DEGR(J))-100.*XSTD(I,MTH)+XMEAN(I,MTH),  
IXMEAN(I,MTH)*XSTD(I,MTH)  
70 PSUM=PSUM+PIVD  
80 PROB(-,MTH)=PSUM/NP  
RETURN  
END
```

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SUBROUTINE FITNC

C FITNC FINDS THE GAUSS-CURVE THAT CLOSEST-FOLLOWS THE UNICUE DISTR.

```

COMMON A,TNMP,STDN,LN3,DIST,I,THI,I,FLO,FTR,MTH,PROB,TPR,REV,
      ITMA,SCAEC,USSA,PRI,PINV,TIM,NIN,DEGR,PLAI,I555,ENAX,EAVR
      DIMENSION A(1117,4,2),TNMP(57,4,2),STDN(57,4,2),DIST(57,4,2),
      IPROB(170,5),IPR(3,5),TMN(5,195)CALC(SRUSSA(5,195)),PR(120,5),
      2PI:V(30,5),TIM(30,5),DEGR(170),LA(LT(5)),EMAX(5),EAVR(5),
      DIMENSION SPKV(13),SIGMA(12),ST(12)
      DATA (SPRV(L),L=1,13)/99865,99379,97725,93319,84134,
      1,69146,0,50130,0,30854,0,15866,0,06681,0,02275,0,00621,0,00135/
      LLOW=1
      LHIGH=12
      L=1
      IF (PROB(I,MTH).LT.SPRV(L)) 630,835
      830 LLow=LLow+1
      L=L+1
      GO TO 625
      835 L=13
      840 IF (PROB(I,MTH).GT.SPRV(L)) 850,860
      850 LHIGH=LHIGH-1
      L=L-1
      GO TO 840
      860 DO 940 L=LLOW,LHIGH
      IF (L.EQ.7) GO TO 940
      J=1
      IF (PROB(J,MTH).GT.SPRV(L)) 910,920
      910 J=J+1
      GO TO 900
      920 ST(L)=DEGR(J)-(SPRV(L)-PROB(J,MTH))/(PROB(J-1,MTH)-
      1PROB(J,MTH))
      940 CONTINUE
      TNM(MTH)=(ST(5)+ST(6)+TPR(1,MTH)+ST(8)+ST(9))*0.2
      SSUM=0.
      DO 1000 L=LLOW,LHIGH
      IF (L.EC.7) GO TO 1000
      SIGMA(L)=2.*ASS((ST(L)-TNM(MTH))/((L-7)))
      SSUM=SSUM+SIGMA(L)
      1000 CONTINUE
      SCALC(MTH)=SSUM/(LHIGH-LLow)

```

C ASSESS THE ACCURACY OF THE CURVE-FITTING, BY INDICATING AN
C ARITHMETIC MEAN- AND A EXTREME DEVIATION FROM THE ROUTE PROB•VALUE
ES = 0.
EMAX(MTH) = 0.
DO 1200 J=1,IIR
PFIT = 1.0 - CNORML(DEGR(0.0-100.*SCALC(MTH)+TMN(MTH)+TAN(MTH)).
1*SCALC(MTH))
EI = ABS(PFIT)-PFIT
IF (EI.GT.EMAX(MTH)) EMAX(MTH) = EI
1200 LS = ES + EI
EAVR(MTH) = ES/IIR
RETURN
END

SUBROUTINE PINT(TINI,LVL)

C PINT CALC. PROBABILITIES ON TEMP. INTERVALS ABOUT THE USSA ITEM

```

COMMON A,TNMP,STDN,LM3,DIST,ITHI,ITLO,ITR,MTH,PROB,TPREV,
1TNAU,SCALC,USSA,PINV,TIM,NINT,DEGR,LAUT,1555,EMAX,EAVR
DIMENSTON A(1127,4,2),TNMP(57,4,21),STDN(57,4,21),DIST(57),
1PROB(170,5),TPR(3,5),TMN(5),SCALC(5),USSA(5),PRI(30,5),
2PIRV(30,5),TIM(30,1),DEGR(70,1),LAUT(5),EMAX(5),EAVR(5),
SIAMESIGN TIL(30),
IF (MTH.GT.1) GO TO 80
C DETERMINE NUMBER OF INTERVALS
NNEG= IFIX((USSA(LVL)-ITLO)/TINT)+1
NTOT= IFIX((ITHI-ITLO)/TINT)+1
80 DO 200 K=1,NTOT
IF (MTH.GT.1) GO TO 110
TIL(K)= USSA(LVL)-TINT*(NNEG+1.5,K)
TIM(K)= TINT*(K-1-NNEG)
110 PRI(K,MTH)= 1.
IF (DEGR(1).GE.TIL(K)) GO TO 300
IF (DEGR(1).LE.TIL(K)) GO TO 260
J=i
170 DIFF= DEGR(J)-TIL(K)
IF (DIFF) 200,210,210
200 J=J+1
GO TO 170
210 PRI(K,MTH)= PROB(J,MTH)+(PROB(J-1,MTH)-PROB(J,MTH))*DIFF
260 PRI(K,MTH)= 0.
300 CONTINUE
NINT= NTOT-1
DO 360 KA=1,NINT
360 PIKV(KA,MTH)= PRI(KA,MTH)-PRI((KA+1,MTH))
RETURN
END

```

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C SUBROUTINE WBIN(A1,A2,M,NTAPE,IRR)

```
DIMENSION A1(70),A2(70),B(16)

NAME=0
CALL WRTEP(A1,1,NAME,M,1,B,C,O,NTAPE,IRR)
IF (IRR>N) GO TO 50
NAME=0
CALL WRTEP(A2,1,NAME,M,1,B,C,O,NTAPE,IRR)
IF (IRR>N) GO TO 50
GO TO 80
50 ENDFILE NTAPE
80 RETURN
END
```

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```

FUNCTION CNORML(XH,XL,XM,XS)      CHIANG E. C. H. 6610004 6600
C   500887  CNORML  PS-497
C   P(X)=5*(1.-(1.-(1.+(1.12821*X+0.88854027*X**2
C   1+.02743349*X**3-.000039646*X**4+.00328975*X**5
C   2)*8))
C   IF NM=1, X1 AND X2 SHOULD BE STANDARDIZED TO NM=1.
C   X1=(XH-XM)/XS
C   X2=(XL-XM)/XS
1000 Z1=X1/1.0414213567
Z2=X2/1.0414213567
IF (Z1*Z2)<1.0203
WHEN Z1 AND Z2 HAVE DIFFERENT SIGN
Z2=ABS(Z2)
CNORML=P(Z2)+P(Z1)
GO TO 100
TO FIND WHETHER Z1 OR Z2 IS 0
CNORML=P(ABS(Z1+Z2))1
GO TO 100
WHEN Z1 AND Z2 HAVE THE SAME SIGN
Z2=ABS(Z2)
Z1=ABS(Z1)
CNORML=ABS(P(Z2)-P(Z1))
100 RETURN
END

```

850301 95 1 0 9
 5 -5 7 9 -1 10 -1 11 -1
 5 1 7 9 -1 10 -1 11 -1
 12 1 13 1 21 1 15 3
 -1 25 47 48 0
 2 1 2
 2 310 80

SEASONAL AND ANNUAL DISTRIBUTION
FOR GIVEN ALT. AND ROUTE

EC 285 180 75
 550 860
 -0- JANUARY
 550 830
 -1- APRIL
 550 800
 -7- JULY
 550 770
 -8- OCTOBER
 550 740
 --- ANNUAL

TEMPERATURE PROBABILITY TO EXCEED

0. 1.
 0 -1 0 0 9
 5 -5 7 9 -1 10 -1 11 -1
 12 1 13 1 21 1 15 3
 -1 25 47 48 0
 2 1 2
 2 310 80
 -0- JANUARY
 550 830
 -1- APRIL
 550 800
 -7- JULY
 550 770

-8 OCTOBER
550 740
-- ANNUAL

TEMP INTERVAL REF. TO STD.
PROBABILITY ON INTERVAL

0.4

0.0

0.0

0.0

0.0

APPENDIX C

INPUT DATA DESCRIPTION AND ARRANGEMENT

The table below describes the preparation of the data input cards.

CARD	LOCATION COLUMN	INPUT ITEM	MODE OF NUMBER	DESCRIPTION
1	1-6	NALT	INT.	Number of altitudes to be used throughout the calculations.
	7-12	ALTI(1)	INT.	First altitude (feet) for which data is to be calculated.
	13-18	ALTI(2)	INT.	Second altitude in calculation. ALTI(2) must be larger than ALTI(1) (altitudes entered in order of increasing numbers). ALTI(2) omitted if NALT equal to one.
2	1-6	OPT(1)	INT.	Option; = 1 if printing of all calculated data, = 0 if not
	7-12	OPT(2)	INT.	Option; = 1 if temp. output in Fahrenheit, = 2 if Celsius
	13-18	OPT(3)	INT.	Option; = 1 if probability on intervals desired, = 0 if not
	19-30	TINT	REAL	Width of temp. intervals, can be omitted if OPT(3) = 0, OPT(2) controls input unit.
3	1-6	ROUTE	INT.	Number of routes to be calculated in the specified manner.
4	1-80	COMMENT	ALPHANUM	Description of route; This card must not be omitted.
5	1-12	GLAT(1)	REAL	Start point latitude in degrees and minutes to the conventional notation; north latitudes positive, south negative.
	13-24	GLNG(1)	REAL	Start points longitude in degrees and numbers to the conventional notation; west longitudes positive, east negative.
	25-36	GLAT(2)	REAL	End point latitude
	37-48	GLNG(2)	REAL	End point longitude

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Input data to process first work is now complete. For additional routes, repeat cards 4 and 5 until specified NROUTE are imputed.

Input data to generate the sample case output shown in figures 5 through 7 are listed below:

COLUMNS

1	6	12	18	24	30	36	48	60	72
1	30,000								
0		1	1	5					
1									
Johannesburg to London				4896 N.M.					
-26.08	-28.15		51.28		.27				

四

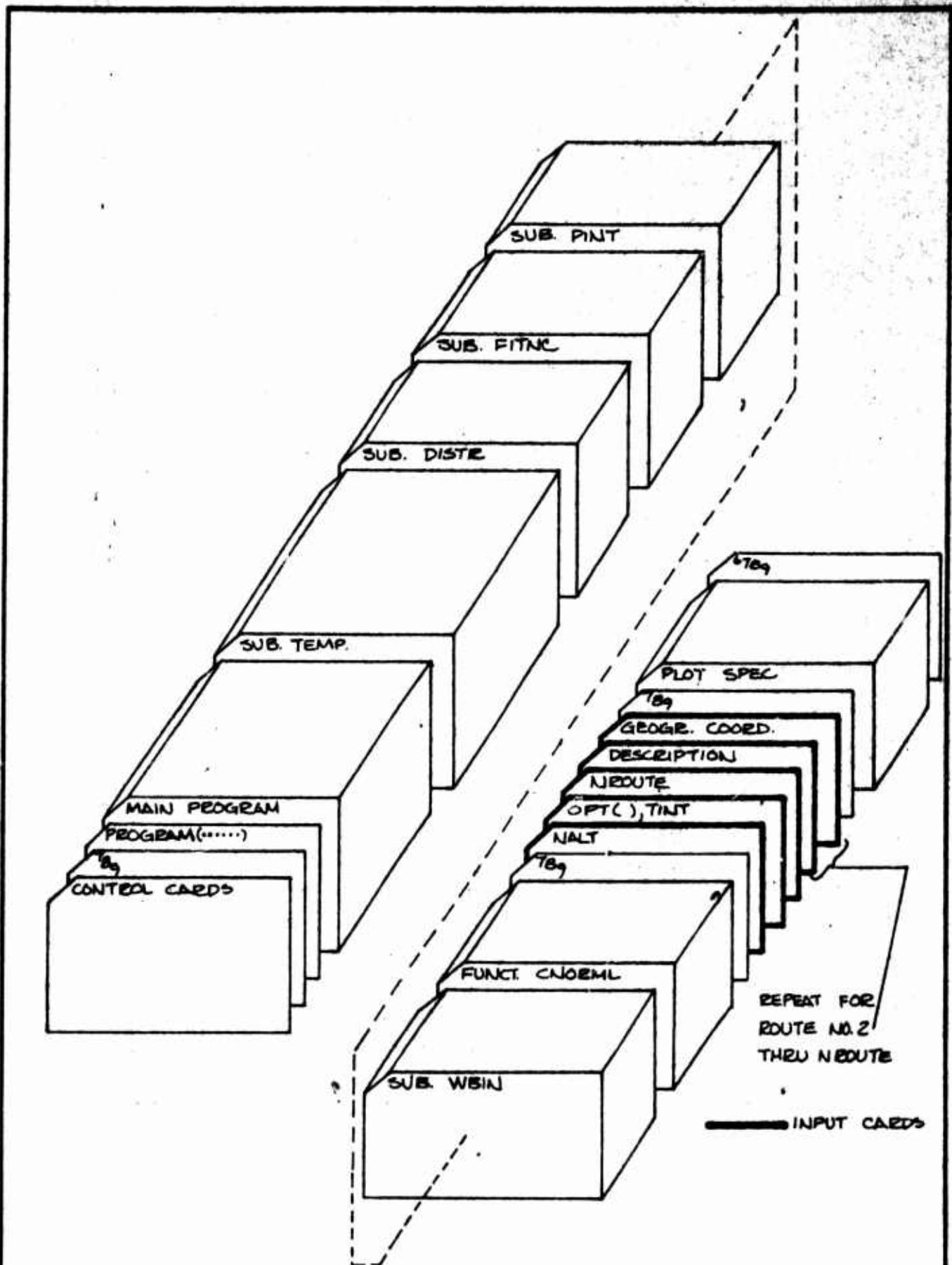
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APPENDIX D - Part 1



CALC			REVISED	DATE	DECK SET-UP	FIG. 19
CHECK						
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Dyn	BFB				THE BOEING COMPANY RENTON, WASHINGTON	

APPENDIX B - Part 2

KV01,136-CW110000.
ACCT 853301 P XM3D1S1R•4D/JOLSSON•K•V.
REQUEST TAPE8.
(66-2453/INPUT)

159-8/655-5517/6-8503-00/8

REWIND (TAPE8)
OFFLINE. S/C 4020 REQUEST
COMMENT. PROJECT FORM NO. 3
COMMENT. CAMERA 16MM
COMMENT. TYPE OF OUTPUT
COMMENT. OPER. INITIALS
REQUEST TAPE99,X.
COPYBF (EOF, TAPE99,3)
REWIND(TAPE99)
RUN(S)
IGC.
UNLOAD (TAPE8)
DROPFIL (TAPE8)
USERFL(ADVSYS,TEL093)
TEL093.
CHKTAPE(TAPE99)
EXIT.
COPYBF (EOF, TAPE99,3)
CHKTAPE(TAPE99)
UNLQDE(TAPE99)

CONTROL CARDS

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APPENDIX E

FORTRAN NOMENCLATURE

The following table gives the nomenclature for the main program and the subprograms. Only the more significant terms are included. The nomenclature for terms, whose values are stored in the COMMON region of the computer, are given first. In the table, dummy subscripts I, J and K are used to denote 1, 2 and 3rd dimension respectively.

NOMENCLATURE FOR THE COMMON STORAGE

FORTRAN	DESCRIPTION	UNITS
SYMBOL		
A (I,J,K)	Meteorological data. Coded in the form <u>aaaa</u> <u>bb</u> , where <u>aaaa</u> is mean-temp. in tenths and <u>bb</u> is standard deviation in tenths also. I denotes location on the earth J denotes month K denotes altitude	C (Celsius)
DEER(I)	Temperatures calculated and used for one altitude	C
DIST(I)	Cumulative distance from start/terminal to point i	N.M.
EAVR(I)	Average error in fitted normal curve	---
EMAX(I)	Maximum error in fitted normal curve	---
ITHI	Upper boundary of temp. range considered for one altitude.	C
ITLO	Lower boundary of temp. range considered for one altitude.	C
ITR	Number of temperature steps used in calculation	---
I55S	Indicator for region of earth where meteorological raw data are not available (68000 ft, south of 55 S.lat.)	---
LALT(I)	Altitude	FEET

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FORTRAN SYMBOL	DESCRIPTION	UNITS
LM3	Number of points i on the route	---
MTH	Months number. 1 through 5 per following denotation. 1 for January 2 for April 3 for July 4 for October 5 for Annual	---
NINT	Number of intervals employed for probability on intervals	---
PINV(I,J)	Probability on individual temp. intervals of given width and median temperature.	N (Normalized)
PRI(I,J)	Cumulative probability value at lower boundary of interval	N
PROB(I,J)	Cumulative probability to exceed temp. values given by DEGR(I)	N
SCALC(I)	Standard deviation for fitted normal distribution curve	C
STDN(I,J,K)	Standard deviation for the points along the route.	C
TIM(I)	Interval median temperature	C
TMN(I)	Mean values for fitted normal distribution curves	C
TNMP(I,J,K)	Mean temp. for the points along the route	C
TPR(I,J)	Temperatures corresponding to .50, .75 and .85 probability <u>not</u> to exceed	C
USSA(I)	United States Standard Atmosphere, 1962	C

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ADDITIONAL NOMENCLATURE FOR THE MAIN PROGRAM AND SUBPROGRAMS
For complete nomenclature to subroutine TEMP, see Document D6-6833TN.

ALTI (I)	Altitudes on which calculations are to be carried out.	Feet
----------	--	------



<u>FORTRAN SYMBOL</u>	<u>DESCRIPTION</u>	<u>UNITS</u>
CCF(Dummy)	Arithmetic statement functions for temperature conversion from Celsius to Fahrenheit.	F
GLAT(1)	Latitudes of terminal points	°.
GLNG(I)	Longitudes of terminal points	°.
IRR	Error indicator for writing on binary tapes	---
LVL	Level (altitude)	FEET
MALT	Number of altitudes used in the calculations	---
MROUTE	Number of routes	---
NTAPE	Binary tape number	---
OPT(I)	Options for calculations	---
PIOD(I)	Array PINV(I,J) transformed to 1-dimension	H
PROD(I)	Array PROB(I,J) transformed to 1-dimension	H
TML(I,J)	Array TEMP(I,J,K) transformed for 2-dim	C
TSDFL(I,J)	Array STDM(I,J,K) transformed to 2-dim.	C

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LIFT OF NOMENCLATURE

<u>Symbol</u>	<u>Description</u>
$F_N(\vartheta)_i$	Cumulative probability that T shall be less than or equal to ϑ , for point "i" of a route.
$F_R(\vartheta)$	Cumulative probability ($T \leq \vartheta$) for the entire route.
$F_{RA}(\vartheta)$	$F_R(\vartheta)$ for the annual case
m_F	Mean value for fitted normal curve.
m_i	Mean value for point "i" of a route.
n	Number of points on a route, terminals included.
σ_F	Standard deviation for fitter normal curve.
σ_i	Standard deviation for point "i" of a route.
t_{50}	Temperature for F_R equal to 50 percent.
t_r	Interpolated temperature used when fitting a normal curve.
T	Function that describes the occurrence of different temperatures ϑ at a given point. In this case a Gaussian function.
ϑ	Independent variable (temperature)

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1	Illustration of Calculation Method, Part 1.	20
2	Illustration of Calculation Method, Part 2.	21
3	Average Temperature at 300 mb in January.	22
4	Standard Deviation of Temperature at 300mb in January.	23
5	Program Printout (Sample case)	24
6	Seasonal and Annual Distribution for Given Altitude and Route (Sample case)	25
7	Probability on Intervals of Specified Width (Sample Case)	26
8	747 A/C Pack Ram Air Drag. Route Temperature Distribution.	27
9	747 A/C Pack Ram Air Drag. Regional Temperature Distribution.	28
10	99, 90, 50, 10 and 1% Probability of Exceeding, Temperature. Continental United States.	29
11	99, 90, 50, 10 and 1% Probability of Exceeding, Temperature. Continental United States combined with North Atlantic, Central and Southern Europe.	30
12	99, 90, 50, 10 and 1% Probability of Exceeding, Temperature. Polar Region of Northern Hemisphere.	31
13	99, 90, 50, 10 and 1% Probability of Exceeding, Temperature. Equatorial Region.	35
14	Seasonal and Annual Distribution for Specified Geographical Region. Continental United States at 30,000 ft.	36
15	Seasonal and Annual Distribution for Specified Geographical Region. Continental United States combined with North Atlantic, Central and Southern Europe at 30,000 ft.	37

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<u>Number</u>	<u>Title</u>	<u>Page</u>
16	Seasonal and Annual Distribution for Specified Geographical Region. Polar region of Northern Hemisphere at 30,000 ft.	38
17	Seasonal and Annual Distribution for Specified Geographical Region. Equatorial Region at 30,000 ft.	39
18	Deviations from Standard Atmosphere.	40
19	Deck Set-Up	67

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II. SUMMARY

Accurate information concerning the statistical distribution of enroute ambient temperatures at airplane cruise altitudes is required for many trade studies and performance calculations, and is particularly important when system performance is highly temperature dependent as, for example, in airplane air-conditioning system studies. This document describes a method of calculating the statistical monthly (January, April, July and October) and annual temperature distribution on any Great Circle route for pressures of 700, 500, 300, 200, 150, 100 and 50 millibars, corresponding approximately to altitudes of 10,000 feet, 18,000 feet, 30,000 feet, 40,000 feet, 45,000 feet, 53,000 feet and 68,000 feet.

The route temperature distributions are generated by a computer program. Inputs to the program are the coordinates of the terminal points. Output is in several forms, namely:

1. The route temperature distribution, showing the probability of exceeding any temperature, given in graphical form with the option of tabular form as well.
2. The percentage of the total time that the temperature lies within discrete intervals of specified median and width, given in graphical form with the option of tabular form as well.
3. The mean and standard deviation of the normal curve which best approximates the actual temperature distribution, and the error associated with the normal curve approximation.

The graphical output is illustrated in Figures 6 and 7 for the Johannesburg to London route at 30,000 feet.

Meteorological data based on records compiled over long periods for a selected global network of points furnish the basis for the program.

Sections of an existing program (Boeing Document D6-6833TN, Program No. TAPO03) are employed, as a subroutine, for the determination of mean temperatures and standard deviations at equidistant points 100-200

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IV. CALCULATION OF TEMPERATURE DISTRIBUTION ON A GREAT CIRCLE ROUTE

A. METEOROLOGICAL DATA

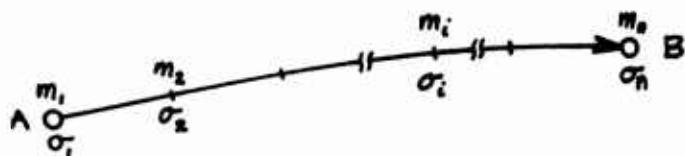
Meteorological data, based on records complied over long periods, have been obtained for a network of 1117 points covering the surface of the earth. The points are located at every latitude which is an exact multiple of five degrees. Within 60° latitude of the equator, the points are located at every longitude which is an exact multiple of ten degrees; on those latitudes which are farther than 60° from the equator the points are located at every longitude which is an exact multiple of twenty degrees. Each pole is represented by one point. The mean and standard deviation of a normal distribution fitted to the actual temperatures over a period of a month are recorded for each point. Data are available for January, April, July and October, these months being assumed to be representative of the seasons; and for pressures of 700, 500, 300, 200, 150, 100 and 50 millibar corresponding approximately to altitudes of 10,000 feet, 18,000 feet, 30,000 feet, 40,000 feet, 45,000 feet, 53,000 feet and 68,000 feet. Thus a total of 62,550 data values are available.

B. GREAT CIRCLE ROUTE CALCULATION

If the geographical coordinates of the terminals of a route are specified, the Great Circle route may be determined by standard methods (Reference 2 for example). The coordinates of most major airports can be found in Reference 1.

C. CALCULATION OF ROUTE TEMPERATURE DISTRIBUTION

If a route is divided into a number of equidistant points, the mean temperature (m_i), and standard deviation (σ_i) at each point (for a given month and altitude) may be obtained by interpolation from adjacent data points (Reference 2). Then if the temperature at any



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Input for the program is described in Appendix C, and sample data for the route Johannesburg to London are included for illustration. It should be noted that exactly antipodal points must not be specified as terminals of a route since such points do not give a uniquely defined Great Circle route. Furthermore, meteorological data are not available on the 700, 500 or the 50 millibar (10,000', 18,000' or 38,000' altitude) levels south of latitude 55 S. A diagnostic is printed if the Great Circle route enters this region on those altitudes.

In order to reduce core storage requirements the DIMENSION statements have been written so that not more than two altitudes can be handled in any one run, and routes are restricted to a maximum length of 11,000 nautical miles.

Output data are in the form of graphs and tables and are self-explanatory. The plots are not labeled by route, but may be identified by the fact that they are generated in the same order that the routes are inputted. Tabular and graphical output corresponding to the input of Appendix C are shown in Figures 5 through 7.

Appendix D shows a complete deck assembly and lists the required control cards. The Fortran nomenclature is shown in Appendix E. The core storage requirement is 110000₈ and the central processor time is approximately equal to

$$CPT = 21.5 + 3.8 \times NALT \times NROUTE \text{ seconds}$$

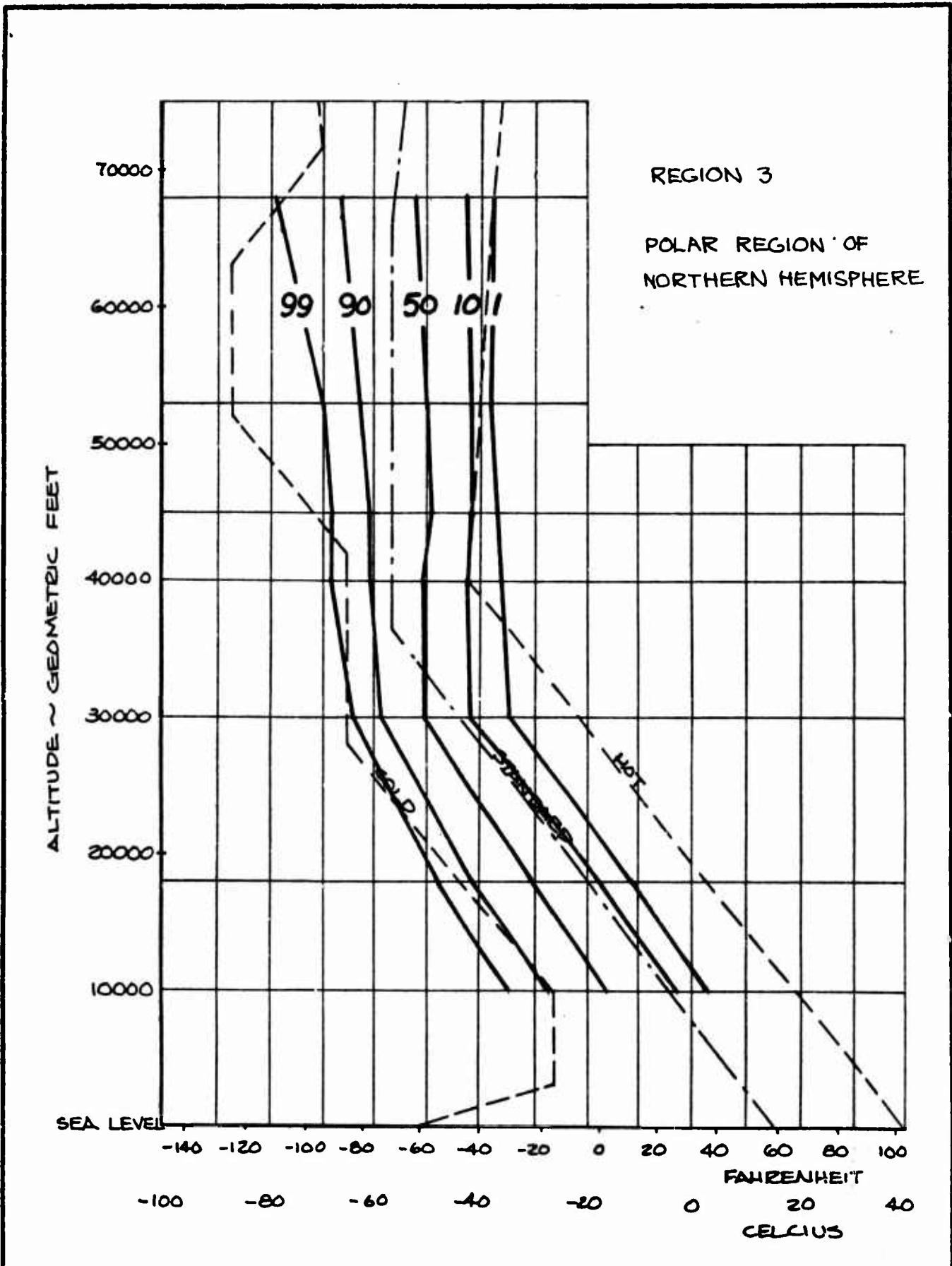
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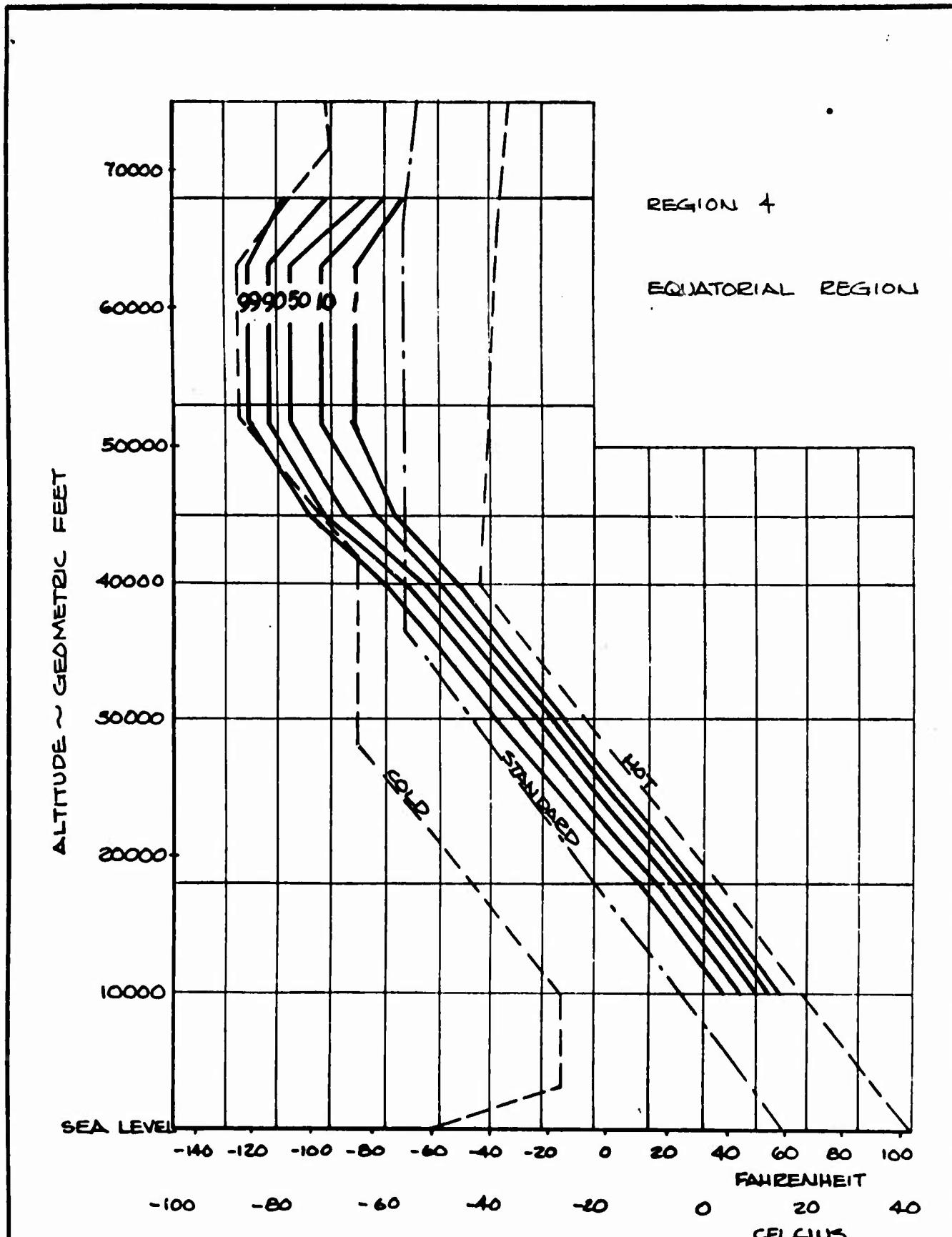
99, 90, 50, 10 AND 1 PER CENT
PROBABILITY OF EXCEEDING
TEMPERATURE

THE BOEING COMPANY
RENTON, WASHINGTON

FIG. 12

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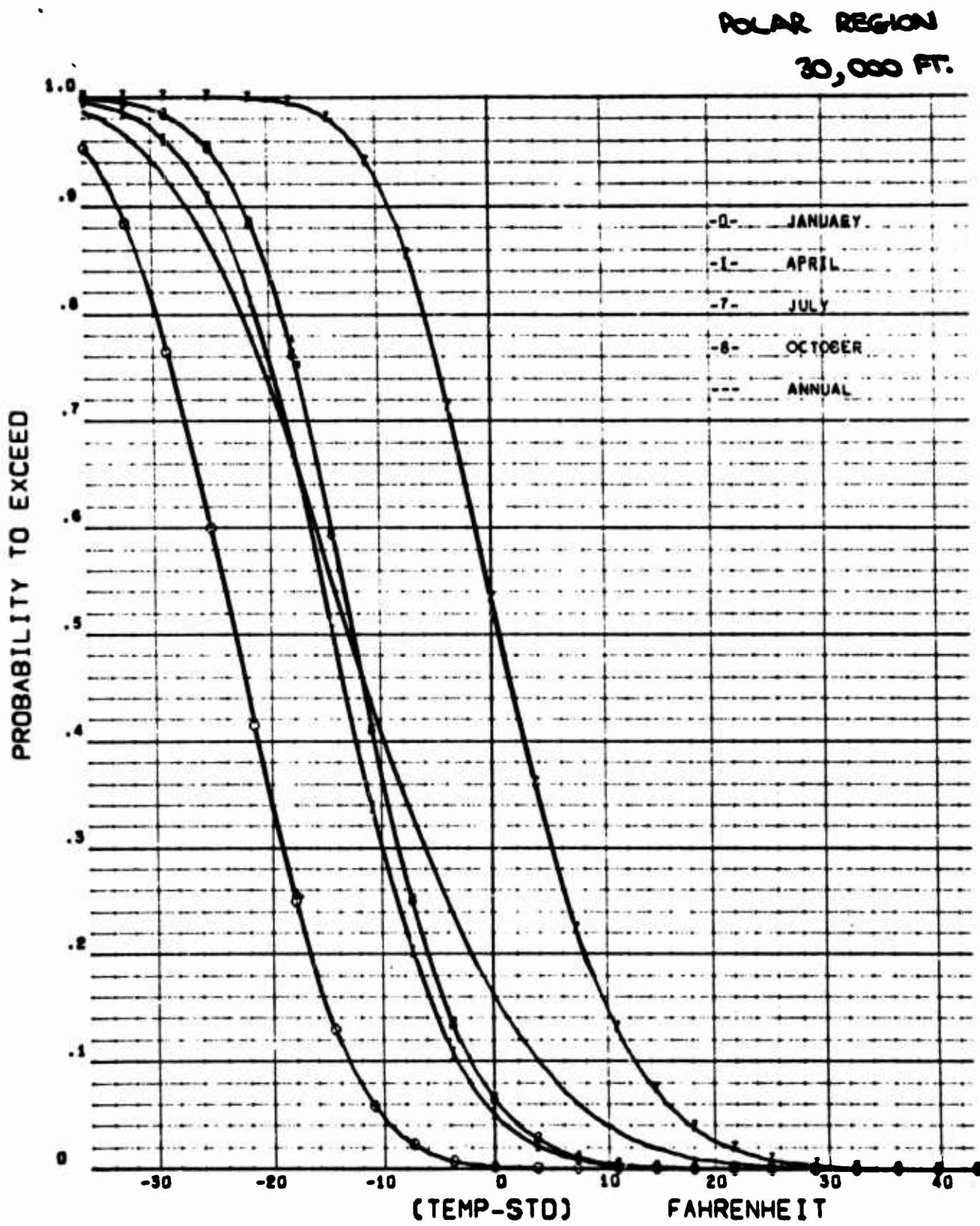
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CALC		REVISED	DATE	99, 90, 50, 10 AND 1 PER CENT PROBABILITY OF EXCEEDING, TEMPERATURE	FIG. 13
CHECK		A			D6-58402
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APPD					PAGE 35
THE BOEING COMPANY RENTON, WASHINGTON					35

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CALC		REVISED	DATE	FIG. 16
CHECK		A		D6-58402
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SEASONAL AND ANNUAL DISTR.
FOR SPEC. GEOGR. AREA

THE BOEING COMPANY
SEATTLE WASHINGTON

Rev. Sym A

B=1.
C=10.

1812 N1=PTNPF(A,B,C,D)
DO 1813 LVL=1,NALT
DO 1813 MTH=1,4
TNMP(IPT,MTH,LVL)=TEMPF(A(N1,MTH,LVL))
1813 STDN(IPT,MTH,LVL)=STDVF(A(N1,MTH,LVL))

2100 CONTINUE

C ACCUMULATED FLIGHT DISTANCE

F1=FLOAT(LGCD)/FLOAT(LM2)
F2=0.0
DO 450 IPT=1,LM3
DIST(IPT)= FLOAT(IRNDF(F2))
F2=F2+F1
450 CONTINUE

C CHECK FOR GEOGRAPHIC REGION WHERE NO RAW DATA AVAILABLE

SLD= 3.1415927*29./36.
DO 2500 LVL=1,NALT
IF (LALT(LVL).LE.-18000.OR.LALT(LVL).EQ.66000) 2400,2500
2400 DO 2500 IPT=1,LM3
IF (RPHI(IPT).GT.SLD) 2420,2500
2420 LSS= 1
PRINT 2460
2460 FORMAT(*METEOROLOGICAL DATA NOT AVAILABLE*,
1* SOUTH OF 55S. LATITUDE*/*SPECIFIED ROUTE ENTERS*,
2* THIS REGION*),
2500 CONTINUE
GO TO 3000
2000 WRITE OUTPUT TAPE 6*2001
2001 FORMAT(1H05X39HAN ERROR OCCURRED IN INTEGER ARITHMETIC)
3000 RETURN
END

APPENDIX D - Part 2

KVOL-T36-CM110000.
ACCT 850301 P XM3
REQUEST TAPE8.
REWIND (TAPE8)
OFFLINE. S/C 4020 REQUEST
COMMENT. PROJECT FORM NO. 3
COMMENT. CAMERA 16MM
COMMENT. TYPE OF OUTPUT
COMMENT. OPER. INITIALS
REQUEST TAPE99,X.
COPYBF(EOF,TAPE99,3)
REWIND(TAPE99,
RUN(S)
RFL(OFF)
MAP(OFF)
LGO.
UNLOAD (TAPE8)
DROPFL (TAPE8)
USERFL(ADVSYS,TEL093)
MAP(OFF)
TEL093.
CHKTAPE(TAPE99)
EXIT.
COPYBF(EOF,TAPE99,3)
CHKTAPE(TAPE99)
UNLOAD(TAPE99)

CONTROL CARDS

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A	Program capability expanded to include the 10,000 feet and 18,000 feet altitude levels.	10/6/69 10/6/69 10/6/69	K.V. Olsson A.J.P. Lloyd J.S. Melchior

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